



WINTER – 17 EXAMINATIONS

Subject Code: **17553**

Model Answer

Page No:

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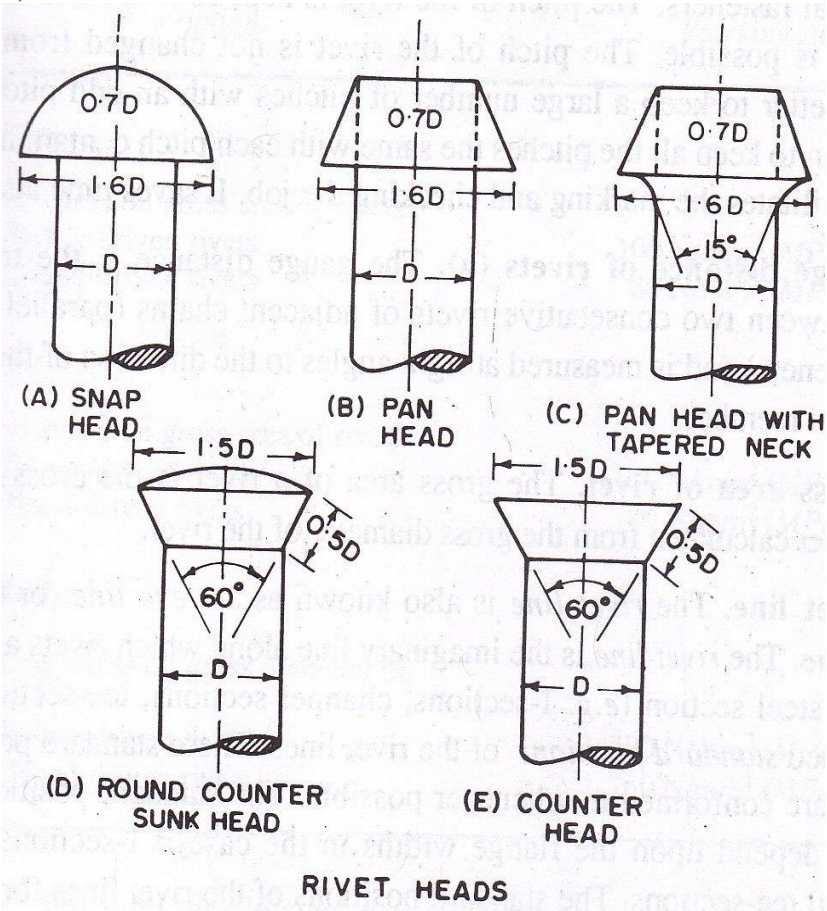
Important Instructions to examiners:

- 1) The answers should be examined by key words and not as word-to-word as given in the model answer scheme.
- 2) The model answer and the answer written by candidate may vary but the examiner may try to assess the understanding level of the candidate.
- 3) The language errors such as grammatical, spelling errors should not be given more importance. (Not applicable for subject English and Communication Skills)
- 4) While assessing figures, examiner may give credit for principal components indicated in the figure. The figures drawn by candidate and model answer may vary. The examiner may give credit for any equivalent figure drawn.
- 5) Credits may be given step wise for numerical problems. In some cases, the assumed constant values may vary and there may be some difference in the candidate's answers and model answer.
- 6) In case of some questions credit may be given by judgment on part of examiner of relevant answer based on candidate's understanding.
- 7) For programming language papers, credit may be given to any other program based on equivalent concept.

Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
1	Attempt any FIVE of the following:		5X4=20
a	<p>Factor of Safety It is defined, in general, as the ratio of the maximum stress to the working stress. Mathematically,</p> <p>Factor of safety = Maximum stress / Working or design stress In case of ductile materials e.g. mild steel, where the yield point is clearly defined, the factor of safety is based upon the yield point stress. In such cases,</p> <p>Factor of safety = Yield point stress / Working or design stress</p> <p>Factors affecting selection of FOS:- 1.The reliability of the properties of the material and change of these properties during service; 2. The reliability of test results and accuracy of application of these results to actual machine parts; 3. The reliability of applied load ; 4. The certainty as to exact mode of failure ; 5. The extent of simplifying assumptions; 6. The extent of localised stresses; 7. The extent of initial stresses set up during manufacture; 8. The extent of loss of life if failure occurs; and 9. The extent of loss of property if failure occurs.</p>	<p>2 marks</p> <p>2 marks. Any 2</p>	4M
b	<p>FG 300- It is a grey cast iron having a minimum tensile strength of 300 N/mm^2</p> <p>40C4:- It is a plain carbon steel having 0.4 % carbon & 0.4% tungsten.</p>	02 mark each	04 marks



c	<p>We know for a square key</p> $w = t = \frac{d}{4}$ <p>Shear strength of key</p> $T = d \times w \times \tau \times \frac{d}{2} \quad \dots \text{--- ①}$ <p>Crushing strength of key</p> $T = d \times \frac{t}{2} \times \sigma_{ck} \times \frac{d}{2} \quad \dots \text{--- ②}$ <p>If the key is equally strong in shearing & crushing</p> $\text{eqn ①} = \text{eqn ②}$ $\therefore d \times w \times \tau \times \frac{d}{2} = d \times \frac{t}{2} \times \sigma_{ck} \times \frac{d}{2}$ $w \times \tau = \frac{t}{2} \times \sigma_{ck}$ $\frac{d}{4} \times \tau = \frac{d/4}{2} \times \sigma_{ck}$ $\frac{d}{4} \times \tau = \frac{d}{8} \times \sigma_{ck}$ $\therefore \boxed{\sigma_{ck} = 2\tau}$		4m
d	<p>Advantages:-</p> <ol style="list-style-type: none">1. The welded structures are usually lighter than riveted structures. This is due to the reason that in welding, gussets or other connecting components are not used.2. The welded joints provide maximum efficiency (may be 100%) which is not possible in case of riveted joints.3. Alterations and additions can be easily made in the existing structures4. As the welded structure is smooth in appearance, therefore it looks pleasing.5. In welded connections, the tension members are not weakened as in the case of riveted joints.6. A welded joint has a great strength. Often a welded joint has the strength of the parent metal itself.7. Sometimes, the members are of such a shape (i.e. circular steel pipe) that they afford difficulty for riveting. But they can be easily welded.8. The welding provides very rigid joints. This is in line with the modern trend of providing rigid frames.9. It is possible to weld any part of a structure at any point. But riveting	2m Any 2	4m

	<p>requires enough clearance.</p> <p>10. The process of welding takes less time than the riveting</p> <p>Disadvantages:-</p> <ol style="list-style-type: none"> 1. Since there is an uneven heating and cooling during fabrication, therefore the member may get distorted or additional stresses may develop. 2. It requires a highly skilled labour and supervision. 3. Since no provision is kept for expansion and contraction in the frame, therefore there is a possibility of cracks developing in it. 4. The inspection of welding work is more difficult than riveting work. 	2M ANY 2	
e	<p>Following are the types of riveted heads:-</p> <ol style="list-style-type: none"> 1) Snap Head 2) Pan Head 3) Pan Head with Taperd Neck 4) Round Countersunk head 5) Counter Head  <p style="text-align: center;">RIVET HEADS</p>	02 m Any 4	4m
f	<p>Bolts of Uniform strength:-</p> <p>If the shank of the bolt is turned down to a diameter equal or even slightly less than the core diameter of the thread (D) as shown in Fig. (b), then shank of the bolt will undergo higher stress. This means that a</p>	2m	4m


$$2m$$

$$\text{diag}$$

g

Perfect frame :

A pin-jointed frame which has got just sufficient number of members to resist the loads without undergoing appreciable deformation in shape is called rigid or perfect frame.

The perfect frame obeys the following condition viz.

$$n = 2j - 3$$

where, n = no. of links and

j= no. of joints

4 marks

4m

2.

Attempt any TWO of the following:

$$8 \times 2 = 16$$

a

Stress Concentration:-

Whenever a machine component changes the shape of its cross-section, the simple stress distribution no longer holds good and the neighbourhood of the discontinuity is different. This irregularity in the stress distribution caused by abrupt changes of form is called stress concentration. It occurs for all kinds of stresses in the presence of fillets, notches, holes, keyways, splines, surface roughness etc.

Causes:- It may occurs due to-

- 1) Change in cross section such as stepped axle, grooves, keyways, threaded holes etc.
- 2) Concentrated load applied at minimum areas of machine parts such as contact between gear teeth.
- 3) Variation in mechanical properties of materials from point to point due to cavities, cracks etc.
- 4) Surface irregularities or poor surface finish.

Remedies:-

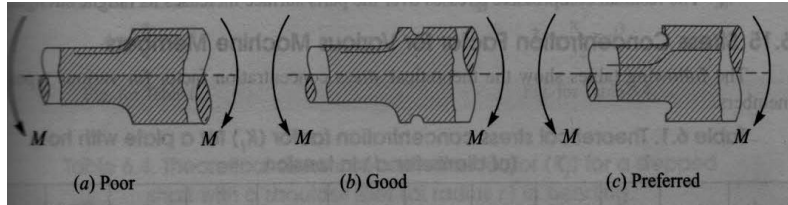
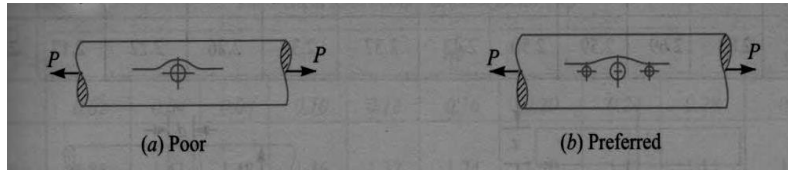
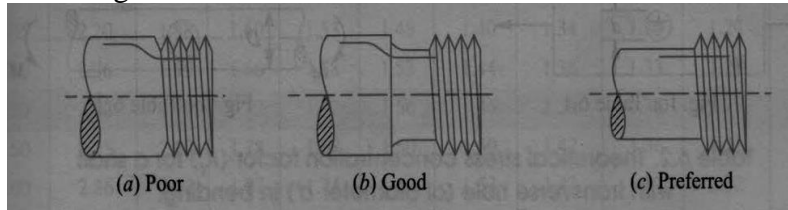
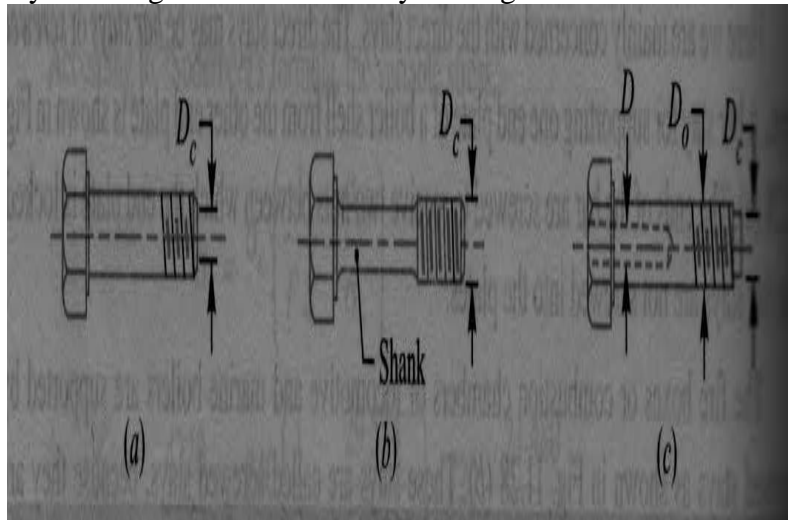
- 1) By fillets, undercutting & notches

 $2m$

8m

2m
(any 2
causes)

4m

	<div data-bbox="354 309 1145 512" data-label="Image">  <p>(a) Poor (b) Good (c) Preferred</p> </div> <p>2) Additional notches & holes</p> <div data-bbox="354 620 1145 788" data-label="Image">  <p>(a) Poor (b) Preferred</p> </div> <p>3) Reducing stress concentration in threaded members</p> <div data-bbox="354 896 1145 1104" data-label="Image">  <p>(a) Poor (b) Good (c) Preferred</p> </div> <p>4) By reducing size of shank or by drilling a hole</p> <div data-bbox="354 1211 1145 1729" data-label="Image">  <p>(a) (b) (c)</p> </div>	<p>(any 2 remedie s with diag)</p>	
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<p>b)</p>	<p> d = diameter of shaft Φ = diameter of hub = $2d$ d_1 = Nominal dia of bolt Φ_1 = diameter of bolt circle = $3d$ n = no of bolts t_f = thickness of flange = $0.5d$ τ_s, τ_b, τ_k = Allowable shear stresses for shaft, bolt & key. τ_c = Allowable shear stress for flange material $\sigma_{ck} \text{ f } \sigma_{cb}$ = Allowable crushing stress for bolt & key. <u>1) Design of hub</u> $T = \frac{\pi}{16} \tau_c \Phi^3 (1 - K_f) \dots \textcircled{1}$ <p>where $K_f = \frac{\Phi}{d}$ here $\Phi = 2d$ & $L = 1.5d$.</p> <p>From equation $\textcircled{1}$ the diameter of hub can be checked If $\tau_c < \tau_{\text{given}}$ design is safe</p> <u>2) Design of key</u> $w = \frac{d}{4}$ $t = \frac{d}{6}$ $l = L = 1.5d$ <u>3) Design for flange</u> $T = \pi \times \Phi \times t_f \times \tau_c \times \frac{\Phi}{2}$ <p>here $t_f = 0.5d$.</p> <p>In above equation if $\tau_c < \tau_{\text{given}}$ design safe</p> </p>	<p>Design of hub 2m</p> <p>Design of key 2m</p> <p>Design of flange 2m</p> <p>Design of bolts 2m</p>	<p>08M</p>
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4) Design of bolts

$$\text{load on each bolt} = \frac{\pi}{4} \times (d_1)^2 \times \tau_b$$

$$\therefore \text{Total load on bolts} = n \times \frac{\pi}{4} (d_1)^2 \times \tau_b$$

\therefore Torque transmitted

$$T = n \times \frac{\pi}{4} (d_1)^2 \times \tau_b$$

from above equation d_1 can be calculated.

checking of bolt under crushing.

$$T = n \times d_1 \times t_f \times \sigma_{cb} \times \frac{D_1}{2}$$

If $\sigma_{cb} < \sigma_{cb \text{ given}}$ design is safe



c)

08
marks

Given

$$W = 80 \text{ mm}, \quad \delta = t = 10 \text{ mm}, \quad \delta t = 70 \text{ N/mm}^2$$

$$\tau = 50 \text{ N/mm}^2 \quad W = 60 \text{ kN} = 60 \times 10^3 \text{ N}$$

To Find the effective length of single transverse fillet weld. i.e. l_1

$$\therefore l_1 = W - 12.5$$

$$= 80 - 12.5$$

$$\boxed{l_1 = 67.5 \text{ mm}}$$

1m(L_1)

Tensile strength of plate (W_{dt}):-

$$W_{dt} = 0.707 \times \delta \times l_1 \times \delta t$$

$$= 0.707 \times 10 \times 67.5 \times 70$$

$$\boxed{W_{dt} = 33.40 \times 10^3 \text{ N}}$$

2m(W_{dt})

Shearing strength of plate (W_τ):-

$$W_\tau = 2 \times 0.707 \times \delta \times l_2 \times \tau$$

$$= 2 \times 0.707 \times 10 \times l_2 \times 50$$

$$\boxed{W_\tau = 707 l_2}$$

2m(W_τ)

We know Total load Carried by plate

$$W = W_{dt} + W_\tau$$

$$60 \times 10^3 = 33.40 \times 10^3 + 707 l_2$$

$$\therefore l_2 = \frac{60 \times 10^3 - 33.40 \times 10^3}{707}$$

$$\boxed{l_2 = 37.62 \text{ mm}}$$

2m(L_2)

For starting & stopping of weld run 12.5 mm is to be added

$$\therefore l_2 = 37.62 + 12.5$$

$$\boxed{l_2 = 50.12 \text{ mm}} \quad \therefore \text{length of Parallel fillet weld.}$$

1m



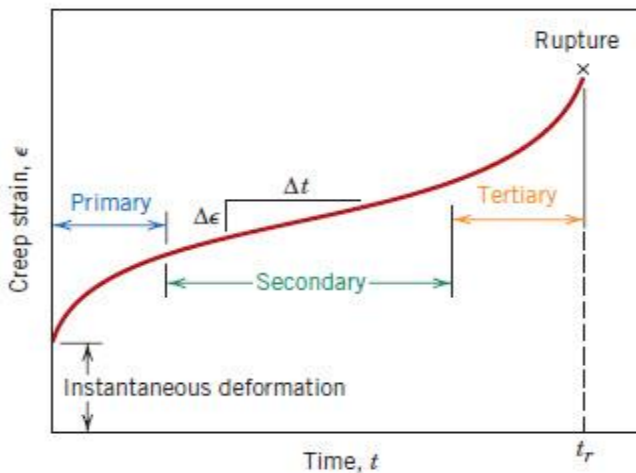
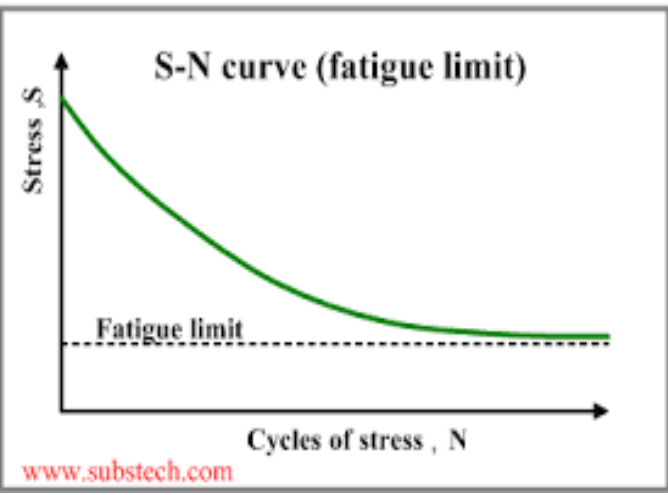
3.	Attempt any TWO of the following:		2X8=16
	<p>1) <u>Thickness of Boiler shell</u>:- It can be determined by using thin Cylindrical Formula</p> $t = \frac{P \cdot D}{2 \sigma_t \times \eta_d} + 1 \text{ mm}$ <p>2) <u>Diameter of Rivet</u> If $t > 8 \text{ mm}$ then $d = 6 \sqrt{t}$ But If $t < 8 \text{ mm}$ then the diameter of rivet can be found by equating shearing resistance with crushing resistance $P_s = P_c$</p> <p>3) <u>Pitch of Rivet</u>:- $P_{\max} = C \cdot t + 41.28 \text{ mm}$ where C = Constant taken from standard table Also pitch can be found by equating $P_t = P_s$ & the least amongst the value is taken.</p> <p>4) <u>Back Pitch (P_b)</u>: $P_b = 0.33 P + 0.67 d$</p> <p>5) <u>Margin (m)</u>: $m = 1.5 d$</p> <p>6) <u>Strap Thickness (t_1)</u>: $t_1 = 0.625 t$</p>	<p>2m thickness</p> <p>2m for dia of rivet</p> <p>2m for pitch or rivet</p> <p>1m back pitch</p> <p>1m for margin</p> <p>Strap thickness is extra data.</p>	<p>08m</p>



b	<p>3b</p> <p><u>Given</u></p> <p>$d_1 = 50 \text{ mm}, d_2 = 37.5 \text{ mm}, W = 12 \times 10^3 \text{ N}$ $d = 400 \text{ mm}, n = 4, \sigma_t = 84 \text{ N/mm}^2$</p> <p>Direct shear load</p> $W_s = \frac{W}{n} = \frac{12 \times 10^3}{4} = 3000 \text{ N}$ <p>The maximum tensile load</p> $W_t = \frac{W \cdot d \cdot d_2}{2 [d_1^2 + d_2^2]} = \frac{12 \times 10^3 \times 400 \times 37.5}{2 [50^2 + 37.5^2]}$ $= \frac{360 \times 10^6}{286.25 \times 10^3} = 1.25 \times 10^3 \text{ N}$ $W_t = 1.25 \times 10^3 \text{ N}$ <p><u>Equivalent load</u></p> $W_{te} = \frac{1}{2} [W_t + \sqrt{(W_t)^2 + 4 (W_s)^2}]$ $= \frac{1}{2} [1.25 \times 10^3 + \sqrt{(1.25 \times 10^3)^2 + 4 (3000)^2}]$ $W_{te} = 3.68 \times 10^3$ <p><u>Size of bolt</u></p> <p>We know</p> $\sigma_t = \frac{W_{te}}{\frac{\pi}{4} (d_c)^2}$ $\therefore 84 = \frac{3.68 \times 10^3}{\frac{\pi}{4} (d_c)^2}$ $d_c^2 \cdot 84 = 4.68 \times 10^3$ $\therefore d_c = 6.61 \text{ mm}$ $d_c \approx 8 \text{ mm}$ <p>We will use bolt of size M10.</p>	<p>02 m for direct shear load</p> <p>2 m for tensile load</p> <p>2m equivalent load</p> <p>2m for size of bolt</p>	8m
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C	<p>3c</p> <p><u>Given</u> $D = 50 \text{ mm}$, $R = 25 \text{ mm}$, $P = 7 \text{ N/mm}^2$, $\sigma_t = 20 \text{ N/mm}^2$ $\sigma_b = 60 \text{ N/mm}^2$</p> <p><u>To Find thickness of Pipe t</u></p> $t = R \left[\sqrt{\frac{\sigma_t + P}{\sigma_t - P}} - 1 \right] = 25 \left[\sqrt{\frac{20 + 7}{20 - 7}} + 1 \right]$ $t = 11.03 \approx 12 \text{ mm}$ <p><u>Outside dia of Flange (D_1)</u></p> $D_1 = D + 2 \times (\text{width of Flange})$ $= 50 + 2 \times 10 \quad \therefore \text{assuming Flange width} = 10 \text{ mm}$ $D_1 = 70 \text{ mm}$ <p>Force Trying to separate the flange</p> $F = \frac{\pi}{4} (D_1)^2 \times P = \frac{\pi}{4} (70)^2 \times 7$ $F = 26943 \text{ N}$ <p>Since the flange is secured by two bolts</p> $\therefore F_b = \frac{F}{2} = \frac{26943}{2} = 13471.5 \text{ N}$ <p>Let d_c = Core dia of bolt</p> <p>we know</p> $F_b = \frac{\pi}{4} (d_c)^2 \times \sigma_b$ $13471.5 = \frac{\pi}{4} \times (d_c)^2 \times 60$ $d_c = 16.9 \approx 17 \text{ mm}$ <p>we know</p> $d_c = 0.84 d_o$ $\therefore d_o = 22 \text{ mm}$ <p><u>Outer diameter of Flange</u></p> $D_o = D + 2t + 4.6d$ $= 50 + 2 \times 12 + 4.6 \times 22$ $D_o = 175.2 \text{ mm} \approx 180 \text{ mm}$ <p><u>Pitch circle diameter</u></p> $D_p = D_o - (3t + 20 \text{ mm})$ $= 180 - (3 \times 12 + 20)$ $D_p = 124 \text{ mm}$	1m given 1m thickness 1m D_1 1m for F 1m for F_b 1m d_o 1m D_o 1m D_p	8M
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4.	Attempt any TWO of the following:		2X8=16
a i)	<p>Creep Curve:- Fig shows an idealized creep curve.</p> <p>When load is applied at the beginning of creep test, Instantaneous deformation occurs.</p> <p>This deformation followed by Creep curve. It occurs in three stages.</p> <ol style="list-style-type: none"> 1) First stage:- it is called primary creep during which creep rate decreases. 2) Second stage:- It is called secondary creep during which creep rate remains constant. This stage occupies major portion of curve. 3) Third Stage:- it is called tertiary creep during which creep rate increases & neck formation will take place finally results in fracture. 	2m	4m
a ii)	<p>Endurance Limit:-</p> <p>Endurance or fatigue limit is defined as the maximum value of completely reversed bending stress, which a standard specimen can withstand without failure for infinite number of cycles of loads.</p> <p>S-N curve:-</p>  <p>Consider a standard mirror polished specimen rotating in a fatigue testing machine & loaded in a bending.</p>	2m	4m

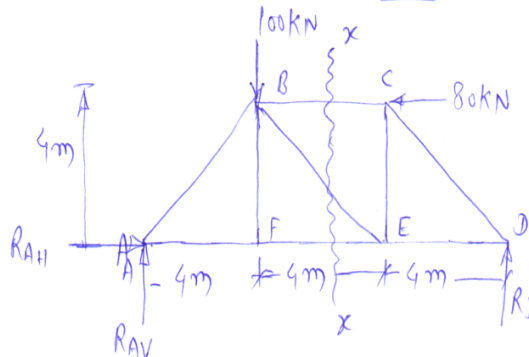


	The specimen is subjected to completely reversed stresses. A record is kept of number of cycles required to produce a failure & results are plotted on Stress-cycle graph as shown in fig.		
b	<p><u>Given,</u></p> $M = 3000 \text{ N}\cdot\text{m} = 3 \times 10^6 \text{ N}\cdot\text{mm}$ $T = 1000 \text{ N}\cdot\text{m} = 1 \times 10^6 \text{ N}\cdot\text{mm}$ $\sigma_{tu} = \sigma_{bu} = 700 \text{ N/mm}^2$ $\tau_{tu} = 500 \text{ N/mm}^2, \quad FOS = 6.$ <p>1) <u>To Find</u> σ_b & τ</p> $\therefore \sigma_b = \frac{\sigma_{bu}}{FOS} = \frac{700}{6} = 116.67 \text{ N/mm}^2$ $\tau = \frac{\tau_{tu}}{FOS} = \frac{500}{6} = 83.33 \text{ N/mm}^2$ <p>2) According to Max Shear Stress Theory Equivalent Twisting Moment</p> $T_e = \sqrt{M^2 + T^2} = \sqrt{(3 \times 10^6)^2 + (1 \times 10^6)^2}$ $T_e = 3.16 \times 10^6 \text{ N}\cdot\text{mm}$ <p>Equating with</p> $T_e = \frac{\pi}{16} \tau d^3 = \frac{\pi}{16} \times 83.33 \times d^3$ $\therefore 3.16 \times 10^6 = 16.36 d^3$ $\boxed{d = 57.80 \text{ mm}} \approx 58 \text{ mm} \approx 60 \text{ mm}$ <p>3) According to Maximum Normal Stress Theory</p> $M_e = \frac{1}{2} [M + \sqrt{M^2 + T^2}] = \frac{1}{2} [3 \times 10^6 + 3.16 \times 10^6]$ $M_e = 3.08 \times 10^6 \text{ N}\cdot\text{mm}$ <p>Equating with</p> $M_e = \frac{\pi}{32} \times \sigma_b \times d^3 = \frac{\pi}{32} \times 116.67 \times d^3$ $\therefore 3.08 \times 10^6 = 11.45 d^3$ $\boxed{d = 64.54 \text{ mm}} \approx 65 \text{ mm}$ <p>\therefore Selecting the larger value $\boxed{d = 65 \text{ mm}}$</p>	<p>given 1m</p> <p>first step 1m</p> <p>Second step 3m</p> <p>Third step 3m</p>	8m

c

46.

Drawing FBD of an entire truss



To Find Support Reactions

$$\begin{aligned} \sum M_A &= 0 \\ + (100 \times 4) - (80 \times 4) - (R_D \times 12) &= 0 \\ 400 - 320 &= 12 R_D \\ R_D &= 6.67 \text{ kN} \end{aligned}$$

$$\begin{aligned} \sum F_x &= 0 \\ R_{AH} - 80 &= 0 \\ R_{AH} &= 80 \text{ kN} \end{aligned}$$

$$\begin{aligned} \sum F_y &= 0 \\ R_{AV} - 100 + R_D &= 0 \\ R_{AV} &= 100 - 6.67 = 93.33 \text{ kN} \end{aligned}$$

Let us consider the equilibrium of truss to the left of section x-x

Taking moment about B.

$$\begin{aligned} (F_{FE} \times 4) + (80 \times 4) &= 93.33 \times 4 \\ F_{FE} &= 13.33 \text{ kN} \end{aligned}$$

Taking moment about E

$$\begin{aligned} (F_{BC} \times 4) + (100 \times 4) &= 93.33 \times 8 \\ F_{BC} &= -86.66 \text{ kN} \end{aligned}$$

$$\begin{aligned} \sum F_x &= 0 \\ 80 - F_{BC} + F_{FE} - F_{BE} \cos 45^\circ &= 0 \end{aligned}$$

$$F_{BE} = -9.43 \text{ kN}$$

Sr No	Member	Magn	Nature
1	FE	13.33	T
2	BC	86.66	C
3	BE	9.43	C

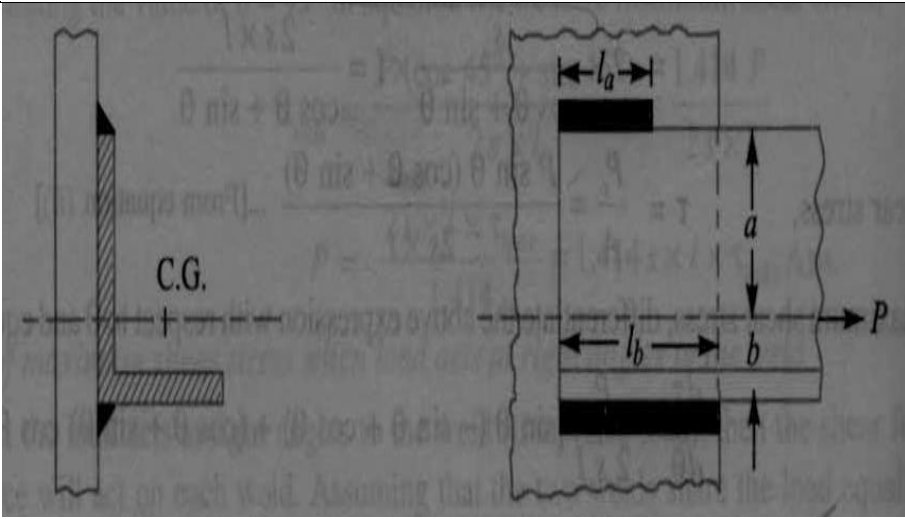
02 m to find support reaction

8m

2m for each member



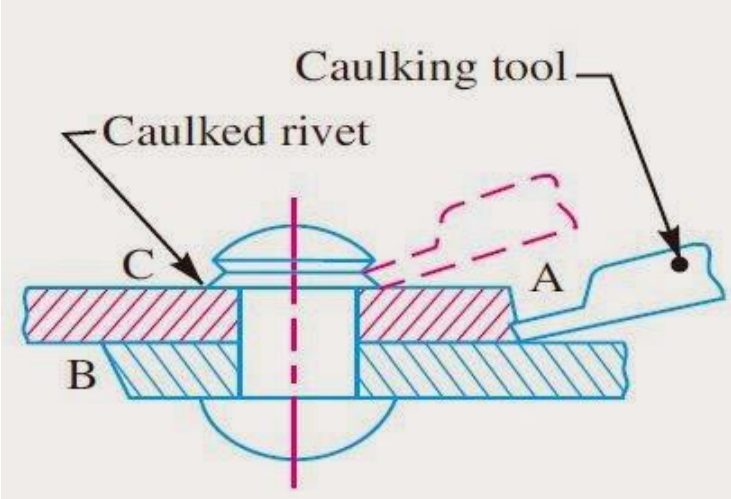
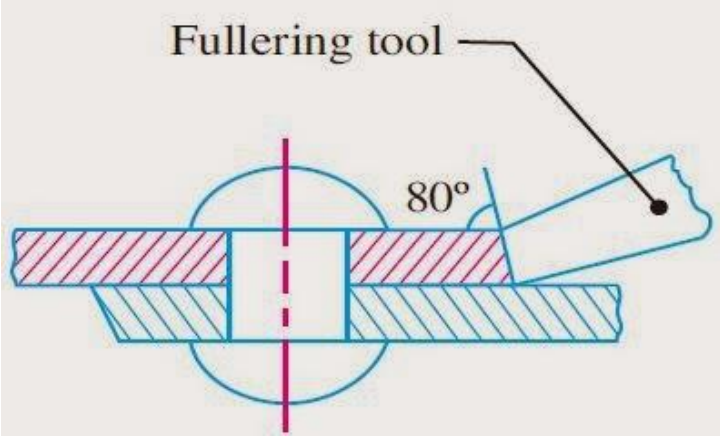
5.	Attempt any TWO of the following:		2X8=16
a	<p><u>5a</u></p> <p><u>Given</u></p> <p>$P = 40 \text{ kW} = 40 \times 10^3 \text{ W}$, $N = 350 \text{ rpm}$</p> <p>$\tau_{\text{shaft}} = 40 \text{ N/mm}^2$, $\tau_{\text{key}} = 40 \text{ N/mm}^2$, $\tau_{\text{muff}} = 15 \text{ N/mm}^2$</p> <p>$\sigma_{\text{ck key}} = 80 \text{ N/mm}^2$</p> <p>1) <u>To Find Torque</u></p> $P = \frac{2\pi NT}{60} \therefore T = \frac{60P}{2\pi N} \therefore T = \frac{60 \times 40 \times 10^3}{2 \times \pi \times 350}$ $\therefore T = 1.09 \times 10^3 \text{ N.m} \quad \boxed{T = 1.09 \times 10^6 \text{ N.mm}}$ <p>2) <u>Diameter of shaft</u></p> $T = \frac{\pi}{16} \tau d^3 \Rightarrow 1.09 \times 10^6 = \frac{\pi}{16} \times 40 \times d^3$ $\therefore \boxed{d = 51.77 \text{ mm} \approx 55 \text{ mm}} \text{ OR } \boxed{d \approx 52 \text{ mm}}$ <p>3) <u>Design of Muff/sleeve:-</u></p> <p>By using empirical relation</p> <p>Outer dia of muff $D = 2d + 13 = 123 \text{ mm}$ OR 117 mm</p> <p>Length of Muff $L = 3.5d = 192.5 = 194 \text{ mm}$ OR 182 mm</p> <p>checking of sleeve under shearing</p> $T = \frac{\pi}{16} \tau D^3 (1 - k^4)$ $1.09 \times 10^6 = \frac{\pi}{16} \tau (123)^3 \cdot (1 - 0.44^4)$ $1.09 \times 10^6 = 351.68 \times 10^3 \tau$ $\therefore \boxed{\tau = 3.10 \text{ N/mm}^2}$ <p>here $\tau < \tau_{\text{given}}$ ie 15 N/mm^2 design is safe</p> <p>4) <u>Design of key</u></p> <p>By using Empirical Relation</p> $w = \frac{d}{4} = \frac{55}{4} = 14 \text{ mm} \quad l = \frac{L}{2} = 97 \text{ mm}$ $t = \frac{d}{6} = \frac{55}{6} = 10 \text{ mm}$	<p>First step 1m</p> <p>Second step 1m</p> <p>Third step 3m</p> <p>Fourth step 3m</p>	8m

	<p>59</p> <p>Shearing of key checking</p> $T = l \times w \times \tau \times \frac{d}{2}$ $1.09 \times 10^6 = 97 \times 14 \times \tau \times \frac{55}{2}$ $\tau = 20.18 \text{ N/mm}^2$ <p>here $\tau < \tau_{\text{given}}$ ie 40 N/mm² design safe</p> <p>checking of key under crushing</p> $T = l \times \frac{t}{2} \times \sigma_{ck} \times \frac{d}{2}$ $1.09 \times 10^6 = 97 \times \frac{10}{2} \times \sigma_{ck} \times \frac{55}{2}$ $\sigma_{ck} = 79.47 \text{ N/mm}^2$ <p>here $\sigma_{ck} < \sigma_{ck \text{ given}}$ ie 80 N/mm² design safe</p>		
b	 <p>Let l_a = Length of weld at the top, l_b = Length of weld at the bottom, l = Total length of weld = $l_a + l_b$ P = Axial load, a = Distance of top weld from gravity axis, b = Distance of bottom weld from gravity axis, and f = Resistance offered by the weld per unit length Moment of the top weld about gravity axis $= l_a \times f \times a$ and moment of the bottom weld about gravity axis $= l_b \times f \times b$ Since the sum of the moments of the weld about the gravity axis must be zero, therefore,</p>	Fig 2m	8M
		Derivation 6m	



	$I_a \times f \times a = I_b \times f \times b$ or $I_a \times a = I_b \times b \dots\dots(i)$ We know that $I = I_a + I_b \dots\dots(ii)$ From equations (i) and (ii), we have $I_a = \frac{I \times b}{(a+b)}$ & $I_b = \frac{I \times a}{(a+b)}$		
C	<p>55</p> <p><u>Isolating Joint A</u></p> <p>Taking $\sum F_y = 0$ $-1 - F_{AB} \sin 45 = 0$ $-1 = F_{AB} \sin 45$ $F_{AB} = -1.41 \text{ kN}$ C</p> <p>Taking $\sum F_x = 0$ $-F_{AC} - F_{AB} \cos 45 = 0$ $F_{AC} = 1 \text{ kN}$ T</p> <p><u>Isolating Joint B</u></p> <p>$\sum F_y = 0$ $F_{BC} + F_{AB} \sin 45 = 0$ $F_{BC} - 1.41 \sin 45 = 0$ $F_{BC} = 1 \text{ kN}$ T</p> <p>$\sum F_x = 0$ $-F_{BE} + F_{AB} \cos 45 = 0$ $F_{BE} = -1 \text{ kN}$ C</p> <p><u>Isolating Joint C</u></p> <p>$\sum F_y = 0$ $-F_{BC} - F_{CE} \sin 45 = 0$ $-1 - F_{CE} \sin 45 = 0$ $F_{CE} = -1.41 \text{ kN}$ C</p> <p>$\sum F_x = 0$ $-F_{CD} - F_{CE} \cos 45 = 0$ $F_{CD} = 1 \text{ kN}$ T</p>	02 Marks for all FBD of isolated joints	8M

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c	<p>Caulking:-</p>  <p>In order to make the joints leak proof or fluid tight in pressure vessels like steam boilers, air receivers and tanks etc. a process known as caulking is employed.</p> <p>In this process, a narrow bunt tool called caulking tool about 5 mm thick and 38 mm in breadth is used. The edge of the tool is ground to an angle of 80°.</p> <p>Fullering:-</p>  <p>A more satisfactory way of making the joints staunch is known as fullering which has largely superseded caulking.</p> <p>In this case, a fullering tool with a thickness at the end equal to that of the plate is used in such a way that the greatest pressure due to the blows occur near the joint, giving a clean finish, with less risk of damaging the plate.</p>	2m	4m
d	<p>following stresses are induced in a bolt, screw or stud when it is screwed up tightly</p> <ol style="list-style-type: none"> 1. Tensile stress due to stretching bolt <p>Since none of the above mentioned stresses are accurately determined, therefore bolts are designed on the basis of direct tensile stress with a large Factor of safety in order to account for the indeterminate stresses. The</p>	4m any 4	4m



	<p>initial tension in a bolt, based on experiments, may be found by the relation $P_i = 2840dN$ P_i = Initial tension in a bolt, and d = Nominal diameter of bolt, in mm.</p> <p>2. Torsional shear stress caused by the frictional resistance of the threads during its tightening The torsional shear stress caused by frictional resistance of the threads during its tightening may be obtained by using the torsion equation. We know that $T/J = T_s/r$ $T_s = T/J \times r = \{ T/(\pi/32) \times d_c^4 \} \times \{ d_c/2 \} = 16 T/ \pi(d_c)^3$ Where T_s = Torsional shear stress, T = Torque applied, and d_c = Minor or core diameter of thread</p> <p>3. Shear stress across the threads. The average thread shearing stress for the screw (T_s) is obtained by using the relation: $T_s = p/(\pi d_c \times b \times n)$ Where b = Width of the thread section at the root. The average thread shearing stress for the nut is $T_n = p/(\pi d \times b \times n)$ Where d = Major diameter.</p> <p>4. Compression or crushing Stress on threads. The compression or crushing stress between the threads (σ_c) may be obtained by using the relation : $\sigma_c = p/ \pi[d^2 - (d_c)^2]n$ Where d = Major diameter, d_c = Minor diameter, and n = Number of threads in engagement.</p> <p>5. Bending stress if the surfaces under the head or nut are not perfectly parallel to the bolt axis. When the outside surfaces of the parts to be connected are not parallel to each other, then the bolt will be subjected to bending action. The bending stress (σ_b) induced in the shank of the bolt is given by $\sigma_b = x.E/2l$ where where x = Difference in height between the extreme corners of the nut or head, l = Length of the shank of the bolt, and E = Young's modulus for the material of the bolt.</p>		
e	<p>Stresses in Pipes: The stresses in pipes due to the internal fluid pressure are determined by Lamé's equation. According to Lamé's equation, tangential stress at any radius x $\sigma_t = \{ [p (r_i)^2] / [(r_o)^2 - (r_i)^2] \} / \{ 1 + [(r_o)^2 / x^2] \}$ And Radial stress at any radius x</p>	4m	4m



	$\sigma_r = \{ [p (r_i)^2] / [(r_o)^2 - (r_i)^2] \} / \{ 1 - [(r_o)^2 / x^2] \}$ <p>where p = Internal fluid pressure in the pipe, r_i = Inner radius of the pipe, and r_o = Outer radius of the pipe</p>		
f	Assumptions in the analysis of truss:- 1) The frame is a perfect one i.e. the relation $n=2j-3$ must be satisfied. 2) All the members are hinged or pin jointed at the ends. 3) The loads are acting only at the joints. 4) The self weight of the member is neglected.	4m 1m each point	4m