



WINTER – 15 EXAMINATIONS

Subject Code: **17304**

Model Answer

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Important Instructions to examiners:

- 1) The answers should be examined by key words and not as word-to-word as given in the model answer scheme.
- 2) The model answer and the answer written by candidate may vary but the examiner may try to assess the understanding level of the candidate.
- 3) The language errors such as grammatical, spelling errors should not be given more importance. (Not applicable for subject English and Communication Skills)
- 4) While assessing figures, examiner may give credit for principal components indicated in the figure. The figures drawn by candidate and model answer may vary. The examiner may give credit for any equivalent figure drawn.
- 5) Credits may be given step wise for numerical problems. In some cases, the assumed constant values may vary and there may be some difference in the candidate's answers and model answer.
- 6) In case of some questions credit may be given by judgment on part of examiner of relevant answer based on candidate's understanding.
- 7) For programming language papers, credit may be given to any other program based on equivalent concept.



Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
<p>1</p> <p>(c)</p>	<p>Rectangular section Bending stress distribution</p> <p>Rectangular section Shear-stress distribution</p>	<p>1</p> <p>1</p>	<p></p> <p>2</p>
<p>(d)</p>	<p>Core-section for rectangular section.</p> $2e_x = \frac{d}{3}$ $\therefore e_x = \frac{d}{6}$ <p>And</p> $2e_y = \frac{b}{3}$ $\therefore e_y = \frac{b}{6}$	<p>2</p>	<p>2</p>

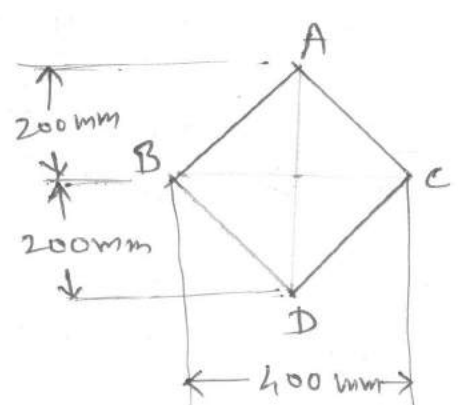


Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
1	<p>(e) The phenomenon of material failure under cyclic loading is called as fatigue.</p> <p>e.g. Machine parts such as axle shafts, springs etc. are generally subjected to cyclic loading conditions.</p>	1 1	2
1	<p>(f) When a homogeneous material is loaded within its elastic limit, the ratio of the lateral strain to the linear strain is constant and is known as poisson's ratio.</p> <p>It is denoted by μ or $\frac{1}{m}$</p>	2	2
1	<p>(g) In thin cylinder, circumferential stresses are developed in the tangential direction to the perimeter (circumference) of the cylinder.</p> <p>As a result of circumferential stress, the cylinder has a tendency to split in two two troughs</p>	$\frac{1}{2}$ $\frac{1}{2}$	



Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
	<p>Longitudinal stresses are developed in the direction parallel to the longitudinal axis of the cylinder</p> <p>As a result of this cylinder has a tendency to split into two small cylinders.</p>	<p>1/2</p> <p>1/2</p>	<p>2</p>
1	<p>(h) Normal stress on an oblique section</p> $\sigma_n = \frac{\sigma_x}{2} (1 + \cos 2\theta) + q \sin 2\theta$ <p>shear stress on an oblique section</p> $\sigma_t = \frac{\sigma_x}{2} \sin 2\theta - q \cos 2\theta$	<p>1</p> <p>1</p>	<p>2</p>
Q.1(B)	<p>(a) Given data: $d = 20 \text{ mm}$, $t = 3 \text{ mm}$ shear stress $q = 180 \text{ MPa}$ $= 180 \text{ N/mm}^2$</p> <p>shearing area of plate, (A) = circumference of hole \times thickness of plate</p>		



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	$A = \pi d \times t$ $= \pi \times 20 \times 3 = 188.495 \text{ mm}^2$ <p>And, Shearing force, $(P) = q \times A$</p> $= q \times \pi d t$ $= 180 \times \pi \times 20 \times 3$ $= 33929.2 \text{ N}$ <p>= Required force to punch the hole is $P = 33929.2 \text{ N}$</p>	1 1 1 1	4
1.B (b)	<p>Here $\Delta ABC = \Delta BCD$</p> <p>$b = 400 \text{ mm}$</p> <p>$h = 200 \text{ mm}$</p>  <p>M.I of square section about its diagonal BC</p> $= \text{M.I of } \Delta ABC \text{ about its base BC}$ $+ \text{M.I of } \Delta CBD \text{ about its base BC}$ $= I_{BC} = \left(\frac{bh^3}{12} \right)_{ABC} + \left(\frac{bh^3}{12} \right)_{CBD}$	1	



Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
	$\therefore I_{Bc} = 2 \times \frac{bh^3}{12}$ $= 2 \times \frac{(400 \times 200^3)}{12}$ $= 5.33 \times 10^8 \text{ mm}^4$	1 2	4
1.B (C)	Given data: $d = 100 \text{ mm}$, $p = 10 \text{ N/mm}^2$ hoop stress $\sigma_c = 120 \text{ N/mm}^2$ Hoop stress, $\sigma_c = \frac{pd}{2t}$ $120 = \frac{10 \times 100}{2 \times t}$ $\therefore t = \frac{10 \times 100}{2 \times 120}$ $= t = 4.166 \text{ mm}$	1 1 2	4
<u>Q.2.</u> (a)	Assumptions of Euler's theory ① The compressive load is exactly axial i.e. it passes through the centroid of the column section.	any four 4 (1 mark for each)	



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	<p>② The material of the column is perfectly homogeneous and isotropic.</p> <p>③ The column is initially straight and of uniform lateral dimensions.</p> <p>④ The column is long and fails due to buckling only.</p> <p>⑤ Shortening of the column due to direct compression is neglected.</p> <p>⑥ The self weight of the column is neglected.</p> <p>⑦ The stress do not exceed the limit of proportionality.</p>		4
2	<p>(b) Given data:</p> <p>Column area, $A = 400 \times 400$ $= 16 \times 10^4 \text{ mm}^2$</p> <p>Area of steel, $A_s = 4 \times \frac{\pi d^2}{4}$ $= 4 \times \frac{\pi (20)^2}{4}$ $= 1256.64 \text{ mm}^2$</p> <p>\therefore Area of concrete,</p> $A_c = A - A_s$ $= 16 \times 10^4 - 1256.64$ $= 158743.36 \text{ mm}^2$		



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2.	$\begin{aligned}\therefore \sigma_s &= 2\sigma_c \\ &= 2 \times 3.1 \\ &= 6.2 \text{ N/mm}^2\end{aligned}$	1	4
(C)	<p>For hollow cylinder,</p> $D = 100 \text{ mm} \quad t = 10 \text{ mm}$ $\therefore \text{Inside dia. } d = 100 - 2t$ $= 100 - 20$ $= 80 \text{ mm}$ $P = 250 \text{ kN} \quad E = 2 \times 10^5 \text{ MPa}$ $= 250 \times 10^3 \text{ N} \quad = 2 \times 10^5 \text{ N/mm}^2$ $\mu = 0.25 \quad L = 800 \text{ mm}$ <p>Area of cylinder,</p> $A = \frac{\pi}{4} (D^2 - d^2)$ $= \frac{\pi}{4} (100^2 - 80^2)$ $= 2827.43 \text{ mm}^2$ <p>\therefore Change in length of cylinder</p> $SL = \frac{PL}{AE}$ $= \frac{250 \times 10^3 \times 800}{2827.43 \times 2 \times 10^5}$ $= 0.353 \text{ mm (increase)}$	1	



Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
	<p>Linear strain, $e = \frac{\delta L}{L} = \frac{0.353}{800}$ $\Rightarrow e = 0.000442$</p> <p>Lateral strain, $e_L = \mu \cdot e$ $= 0.25 \times 0.000442$ $= 0.0001105$</p> <p>also $e_L = \frac{\delta D}{D}$ $\therefore 0.0001105 = \frac{\delta d}{100}$ $\Rightarrow \delta D = 0.01105 \text{ mm (decrease)}$</p>		
2 d	<p>Diameter, $d = 30 \text{ mm}$</p> <p>Fall in temperature, $t = 70 - 20$ $= 50^\circ \text{C}$</p> <p>Length of bar $L = 10 \text{ m}$ $= 10000 \text{ mm}$</p> <p>$\alpha = 12 \times 10^{-6} / ^\circ \text{C}$, $E = 2 \times 10^5 \text{ N/mm}^2$</p> <p>$\therefore$ Temperature stress, $\sigma = \alpha t E$ $= 12 \times 10^{-6} \times 50 \times 2 \times 10^5$ $= 120 \text{ N/mm}^2 \text{ (Tensile)}$</p>		4



Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
	<p>Reactions at the clamps,</p> $P = \sigma \cdot A$ $= \sigma t E \cdot \left(\frac{\pi d^2}{4} \right)$ $= 120 \times \frac{\pi (30)^2}{4}$ $= 84823 \text{ N}$ $= 84.823 \text{ kN}$	1	4
2. e	<p>For the equilibrium of entire bar, ($\sum F_x = 0$), $-120 + 200 - P + 150 = 0$ $\therefore P = 230 \text{ kN}$</p> <p>F.B.D's for each section:</p> <p>Here section AB & CD are under tension and section BC is under compression.</p> <p>Now $\delta L_1 = \frac{P_1 L_1}{A_1 E}$</p>	1	

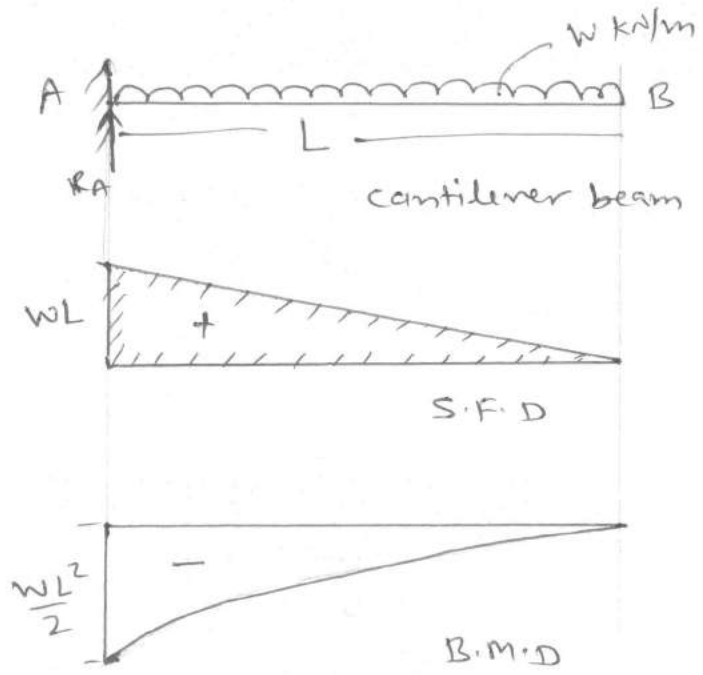


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	$\therefore \delta L_1 = \frac{120 \times 10^3 \times 1000}{30 \times 30 \times 2 \times 10^5}$ $= 0.666 \text{ mm}$	$\frac{1}{2}$	
	$\delta L_2 = - \frac{P_2 L_2}{A_2 E}$ $= - \frac{80 \times 10^3 \times 1000}{20 \times 20 \times 2 \times 10^5}$ $= -1 \text{ mm}$	$\frac{1}{2}$	
	$\& \delta L_3 = \frac{P_3 L_3}{A_3 E}$ $= \frac{150 \times 10^3 \times 1000}{40 \times 40 \times 2 \times 10^5}$ $= 0.468 \text{ mm}$	$\frac{1}{2}$	
	<p>\therefore Net change in the length</p> $\delta L = 0.666 - 1 + 0.468$ $= 0.134 \text{ mm}$ <p>(+ve sign indicates increase in the length of member ABCD)</p>	$\frac{1}{2}$	4



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2. (f)	<p>$\sigma_x = 60 \text{ N/mm}^2$, $\sigma_y = 30 \text{ N/mm}^2$ $\theta = 20^\circ$</p> <p>Normal stress on plane BE ,</p> $\sigma_n = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\theta$ $= \frac{60 + 30}{2} + \frac{60 - 30}{2} \cos(2 \times 20)$ $= 45 + 15 \cos 40$ $= 56.49 \text{ N/mm}^2$ <p>Tangential stress on plane BE ,</p> $\sigma_t = \frac{\sigma_x - \sigma_y}{2} \sin 2\theta$ $= \frac{60 - 30}{2} \times \sin(2 \times 20)$ $= 15 \times \sin 40$ $= 9.641 \text{ N/mm}^2$	<p>1/2</p> <p>1</p> <p>1</p>	



Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
	<p>Angle of obliquity ,</p> $\phi = \tan^{-1} \frac{\delta t}{\delta n}$ $= \tan^{-1} \frac{9.64}{56.49}$ $\therefore \phi = 9.68^\circ$	<p>1/2</p> <p>1/2</p>	<p>4</p>
Q.3	<p>(a)</p>  <p>The diagram shows a cantilever beam AB of length L fixed at A and free at B. A uniformly distributed load w kN/m is applied downwards. Below the beam are the Shear Force Diagram (S.F.D) and Bending Moment Diagram (B.M.D). The S.F.D is a triangle starting at WL at A and ending at 0 at B. The B.M.D is a parabolic curve starting at $\frac{wL^2}{2}$ at A and ending at 0 at B.</p>	<p>1</p> <p>1</p>	



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3. b	<p>Support Reaction:</p> $\sum F_y = 0 \Rightarrow R_A - WL = 0$ $\Rightarrow R_A = WL = 50 \text{ kN}$ <p>Shear force calculations:</p> <p>S.F at A, $F_A = R_A = 50$</p> <p>S.F at B, $F_B = 0$</p> <p>Bending moment calculations:</p> <p>B.M. at A is maximum</p> <p>i.e. $M_A = M_{\max}$</p> $\therefore M_A = -\frac{WL^2}{2} \text{ (Hogging)}$ <p>SFD</p> <p>BMD</p>	1 1 1	4



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	<p>Support Reaction calculations :</p> $\sum F_y = 0,$ $R_A - 80 - 120 + R_B = 0$ $\therefore R_A + R_B = 200 \quad \text{--- (1)}$ $M_A = 0,$ $(80 \times 2) + (120 \times 6) = 8 R_B$ $160 + 720 = 8 R_B$ $\therefore R_B = 110 \text{ kN}$ <p>put this value in eqⁿ (1),</p> $\therefore R_A + 110 = 200$ $\therefore R_A = 90 \text{ kN}$ <p>Shear Force calculations :</p> $F_{AL} = 0 \quad F_{AR} = 0$ $F_{CL} = 90 - 80 \quad F_{CR} = 10 \text{ kN}$ $= 10 \text{ kN}$ $F_{DL} = 10 \text{ kN} \quad F_{DR} = 10 - 120$ $= -110 \text{ kN}$ $F_{BL} = -110 \text{ kN} \quad F_{BR} = -110 + 110$ $= 0$ <p>Bending Moment calculations :</p> <p>B.M at points A & B are zero. i.e. $M_A = 0$ & $M_B = 0$</p> <p>B.M at 'c' $M_c = (90 \times 4) - (80 \times 2)$</p>	1	4

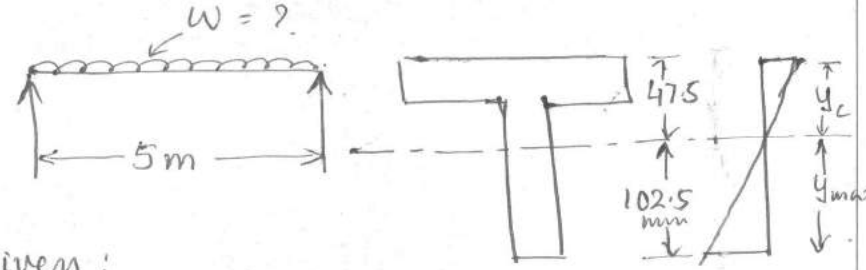


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3e	<p>$\sum M_A = 0$</p> $(15 \times 1) + (30 \times 4) - 5R_B = 0$ $15 + 120 = 5R_B$ $\Rightarrow R_B = 27 \text{ kN}$ <p>put in eqn (1),</p> $R_A + 27 = 45$ $\Rightarrow R_A = 18 \text{ kN}$ <p>SF calculations:</p> $F_{AL} = 0 \quad F_{AR} = 18 \text{ kN}$ $F_{CL} = 18 \text{ kN} \quad F_{CR} = 18 - 15 = 3 \text{ kN}$	1	1

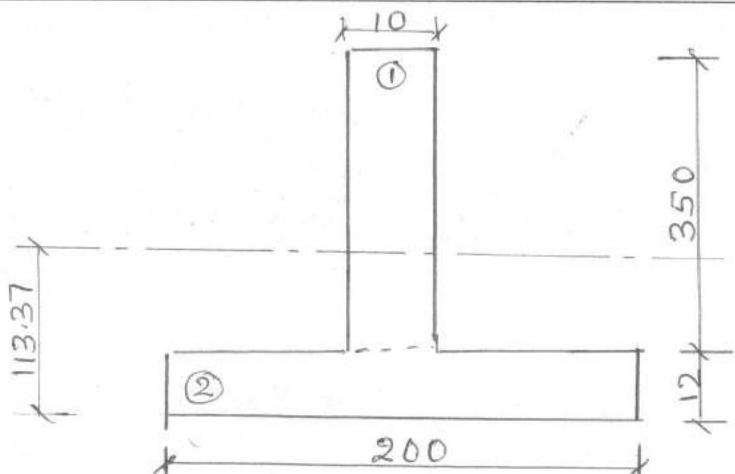


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4. (a)	<p>Given :</p> $D = 100 \text{ mm}, S = 5 \text{ kN} = 5 \times 10^3 \text{ N}$ $\text{Area } A = \frac{\pi}{4} D^2 = \frac{\pi}{4} \times 100^2 = 7853.98 \text{ mm}^2$ $q_{av} = \frac{S}{A} = \frac{5 \times 10^3}{7853.98} = 0.6366 \text{ N/mm}^2$ $q_{max} = \frac{4}{3} q_{av} = \frac{4}{3} \times 0.6366 = 0.8488 \text{ N/mm}^2$ <div style="border: 1px solid black; padding: 5px; width: fit-content; margin: 5px 0;">$q_{max} = 0.8488 \text{ N/mm}^2$</div> <div style="border: 1px solid black; padding: 5px; width: fit-content; margin: 5px 0;">$q_{min} = 0 \text{ N/mm}^2$</div> <p style="text-align: center;"><u>Shear stress dist dia</u></p>		



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4 (b)	 <p>Given :</p> $\sigma_{max} = 160 \text{ MPa} = 160 \text{ N/mm}^2$ $l = 5 \text{ m}$ $I_{xx} = 45 \times 10^6 \text{ mm}^4, C_{xx} = \bar{y}_{top} = 47.5 \text{ mm}$ $\bar{y}_{base} = \bar{y}_{max} = 150 - 47.5 = 102.5 \text{ mm}$ <p>Using bending eqⁿ.</p> $\frac{M_{max}}{I_{xx}} = \frac{\sigma_{max}}{y_{max}}$ $\therefore M_{max} = \frac{\sigma_{max} \times I_{xx}}{y_{max}} = \frac{160 \times 45 \times 10^6}{102.5}$ $= 70.243 \times 10^6 \text{ N-mm}$ $M_{max} = \frac{wl^2}{8}$ <p>Taking w in N/mm.</p> $70.243 \times 10^6 = \frac{w(5 \times 10^3)^2}{8}$ $w = 22.478 \text{ N/mm} = 22.478 \text{ kN/m}$	1/2 1 1/2 1	4

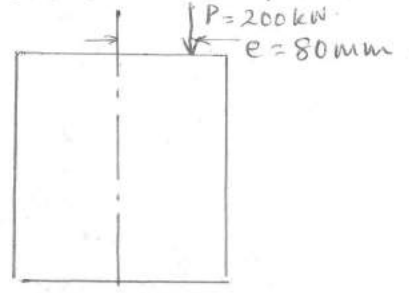
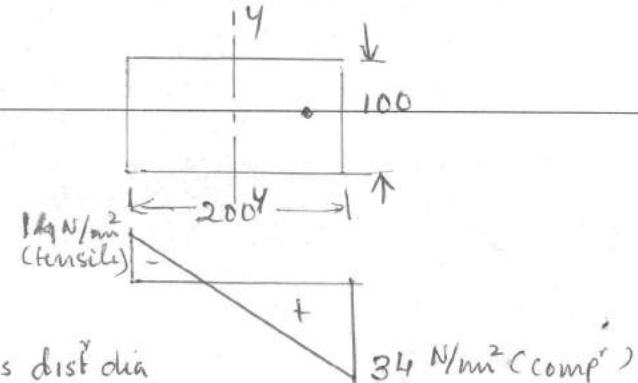


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4. (C)	 <p>Given:</p> $a_1 = 350 \times 10 = 3500 \text{ mm}^2 \quad a_2 = 200 \times 12 = 2400 \text{ mm}^2$ $y_1 = 12 + \frac{350}{2} = 187 \text{ mm} \quad y_2 = \frac{12}{2} = 6 \text{ mm}$ $a = a_1 + a_2 = 5900 \text{ mm}^2$ $\bar{y}_{\text{bottom}} = \frac{a_1 y_1 + a_2 y_2}{a} = \frac{(3500 \times 187) + (2400 \times 6)}{5900} = 113.37 \text{ mm}$ $h_1 = y_1 - \bar{y} = 187 - 113.37 = 73.63 \text{ mm}$ $h_2 = \bar{y} - y_2 = 113.37 - 6 = 107.37 \text{ mm}$ $I_{xx_1} = \frac{b_1 d_1^3}{12} + a_1 h_1^2 = \frac{10 \times 350^3}{12} + 3500 (73.63)^2$ $I_{xx_1} = 54703985.82 \text{ mm}^4$ $I_{xx_2} = \frac{b_2 d_2^3}{12} + a_2 h_2^2 = \frac{200 \times 12^3}{12} + 2400 (107.37)^2$ $I_{xx_2} = 27696760.56 \text{ mm}^4$ $\therefore I_{xx} = I_{xx_1} + I_{xx_2} = 82400746.38 \text{ mm}^4$ $\boxed{I_{xx} = 82.4 \times 10^6 \text{ mm}^4}$	1 1 1	4

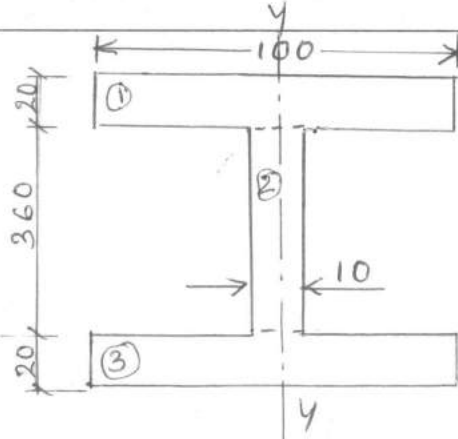
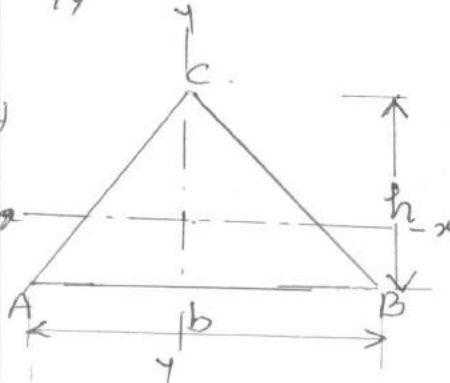


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4. (d)	<p>$a_1 = 8 \times 40 = 320 \text{ mm}^2 = a_3$, $a_2 = 90 \times 10 = 900 \text{ mm}^2$ $y_1 = 10 + \frac{40}{2} = 30 \text{ mm} = y_3$, $y_2 = 10/2 = 5 \text{ mm}$ $h_1 = y_1 - \bar{y} = 30 - 15.39 = 14.61 \text{ mm} = h_3$ $h_2 = \bar{y} - y_2 = 15.39 - 5 = 10.39 \text{ mm}$ $A = a_1 + a_2 + a_3 = 1540 \text{ mm}^2$ $\bar{y} = \frac{a_1 y_1 + a_2 y_2 + a_3 y_3}{A} = \frac{2(320 \times 30) + (900 \times 5)}{1540} = 15.39 \text{ mm}$ $I_{xx_1} = I_{xx_3} = \frac{b_1 d_1^3}{12} + a_1 h_1^2$ $I_{xx_1} = I_{xx_3} = \frac{8 \times 40^3}{12} + 320(14.61)^2 = 110971.339 \text{ mm}^4$ $I_{xx_2} = \frac{b_2 d_2^3}{12} + a_2 h_2^2 = \frac{90 \times 10^3}{12} + 900(10.39)^2$ $I_{xx_2} = 104656.89 \text{ mm}^4$ $I_{xx} = 2(I_{xx_1}) + I_{xx_2} = 2(110971.34) + 104656.89$ $I_{xx} = 32.66 \times 10^4 \text{ mm}^4$ $k_{xx} = \sqrt{\frac{I_{xx}}{A}} = \sqrt{\frac{32.66 \times 10^4}{1540}} = 14.56 \text{ mm}$</p>	1 1 1 1	4

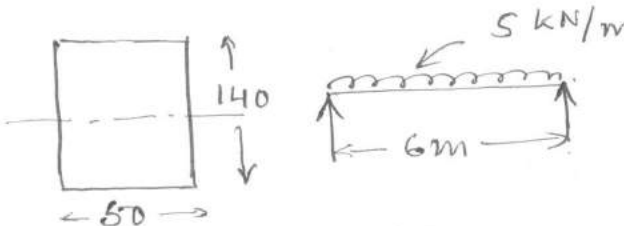


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5 (c)	<p>For no tension condⁿ $D = 40 \text{ mm}, A = \frac{\pi}{4} D^2, \sigma_0 = \sigma_b, \frac{P}{A} = \frac{M}{Z} = \frac{P \cdot e}{Z}$</p> <p>Limit of eccentricity $e' = \frac{Z}{A}$</p> $Z = \frac{I}{y} = \frac{\pi/64 D^4}{D/2} = \frac{\pi}{32} D^3$ $\therefore e = \frac{Z}{A} = \frac{\pi/32 D^3}{\pi/4 D^2} = D/8 = \frac{40}{8} = \underline{\underline{5 \text{ mm}}}$	1 1/2 1/2	4
5 (d)	<p>$P = 200 \text{ kN}, e = 80 \text{ mm}, b = 200 \text{ mm}, d = 100 \text{ mm}$ $A = b \times d = 2 \times 10^4 \text{ mm}^2$</p> <p>Direct stress $\sigma_0 = \frac{P}{A} = \frac{200 \times 10^3}{2 \times 10^4} = 10 \text{ N/mm}^2$</p> <p>Bending stress $\sigma_b = \frac{M}{Z} = \frac{P \cdot e}{Z} = \frac{6 P \cdot e}{d b^2}$</p> $\therefore \sigma_b = \frac{6 \times 200 \times 10^3 \times 80}{100 \times 200^2} = 24 \text{ N/mm}^2$ <p>$\sigma_{\text{max}} = \sigma_0 + \sigma_b = 10 + 24 = 34 \text{ N/mm}^2$ (Compⁿ) $\sigma_{\text{min}} = \sigma_0 - \sigma_b = 10 - 24 = -14 \text{ N/mm}^2$ (Tensile)</p>   <p>Stress dist dia</p>	1 1 1	4



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4. (e)	$I_{yy} = I_{yy_1} + I_{yy_2} + I_{yy_3}$ $= 2\left(\frac{d_1 b_1^3}{12}\right) + \frac{d_2 b_2^3}{12}$ $= 2\left(\frac{20 \times 100^3}{12}\right) + \left(\frac{360 \times 10^3}{12}\right)$ $\frac{I}{yy} = 3363333.333 = 33.63 \times 10^5 \text{ mm}^4$ 	1 1 1 1	4
4. (f) (i)	 $I_{\text{base}} = \frac{bh^3}{12}$ $\frac{I}{xx} = \frac{bh^3}{36}$ <p>or</p> $I_{yy} = \frac{hb^3}{48}$	1 1	
4. (f) (ii)	<p><u>Parallel Axis Theorem:</u></p> <p>The moment of inertia of a plane section about any axis parallel to the centroidal axis is equal to the ^{sum of} moment of inertia of the section about centroidal axis and product of area of the section and square of distance between two axes.</p> <p><u>Perpendicular Axis Theorem:</u></p> <p>If I_{xx} and I_{yy} are the moment of inertia of plane section about two mutually perpendicular axes, then moment of inertia I_{zz} about the third axis z-z, perpendicular to the plane & passing through the intersection of x-x & y-y is given by</p> <div style="border: 1px solid black; padding: 5px; width: fit-content; margin: 10px auto;"> $I_{zz} = I_{xx} + I_{yy}$ </div>	1	4



Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
5 (a)	<p>$D = 300 \text{ mm}, t = 50 \text{ mm}, P = 200 \times 10^3 \text{ N}$ $d = D - 2t = 200 \text{ mm}$ For no tension condition, $\sigma_0 \geq \sigma_b, \frac{P}{A} = \frac{M}{Z} = \frac{Pe}{Z}$ $e_{\max} = \frac{Z}{A}$ $Z_{xx} = Z_{yy} = \frac{\pi}{32} \frac{(D^4 - d^4)}{D} = \frac{\pi}{32} \frac{(300^4 - 200^4)}{300}$ $Z = 2127120.026 \text{ mm}^3$ $A = \frac{\pi}{4} (D^2 - d^2) = \frac{\pi}{4} (300^2 - 200^2) = 39270 \text{ mm}^2$ $\therefore e_{\max} = \frac{Z}{A} = 54.17 \text{ mm}$</p>	1 1 1 1	4
5 (b)	<p> $M_{\max} = \frac{wl^2}{8} = \frac{5 \times 10^3 (6)^2}{8} = 22,500 \text{ N}\cdot\text{m}$ $= 22500 \times 10^3 \text{ N}\cdot\text{mm}$ $I_{xx} = \frac{bd^3}{12} = \frac{50 \times 140^3}{12} = 11.43 \times 10^6 \text{ mm}^4$ $y_{\max} = d/2 = \frac{140}{2} = 70 \text{ mm}$ Using bending eqⁿ $\rightarrow \frac{M_{\max}}{I} = \frac{\sigma_b}{y_{\max}}$ $\therefore \sigma_{b_{\max}} = \frac{M}{I} \times y = \frac{22.5 \times 10^6}{11.43 \times 10^6} \times 70$ $\sigma_{b_{\max}} = 137.80 \text{ N/mm}^2$</p>	1 1 1/2 1/2 1	4



Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
5. f		2 2	4
6 (a)	<p>Given :</p> <p>$D = 100 \text{ mm}$, $L = 2.7 \text{ m}$, $T = 30 \text{ kN-m} = 30 \times 10^6 \text{ N-m}$</p> <p>$G = 75 \text{ GPa}$.</p> <p>Find - τ_{max}.</p> <p>Using relation $T = \frac{\pi}{16} \cdot \tau \cdot D^3$</p> $30 \times 10^6 = \frac{\pi}{16} \times \tau \times 100^3$ <p>or $\tau = 152.79 \text{ N/mm}^2$</p>	1 1 2	4



Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
6. (b)	<p><u>Assumptions in theory of pure torsion :-</u></p> <ol style="list-style-type: none">1. The material of shaft is homogeneous and isotropic2. Stresses are within elastic limit i.e. shear stress is proportional to shear strain.3. Cross sections which are plane before applying twisting moment remain plane after its application i.e. no warping takes place.4. Twist along the shaft is uniform.5. All diameters of the cross section of the shaft remain straight before and after the twist.	Any 4. (1M for each)	4
6. (c)	<p>Given: $T = 24 \text{ kN}\cdot\text{m} = 24 \times 10^6 \text{ N}\cdot\text{mm}$ $d = 0.6D$, $\tau = 80 \text{ MPa} = 80 \text{ N/mm}^2$</p> <p>Using relation $\rightarrow T = \frac{\pi}{16} \left(\frac{D^4 - d^4}{D} \right) \tau$</p> <p>$\therefore 24 \times 10^6 = \frac{\pi}{16} \left(\frac{D^4 - (0.6D)^4}{D} \right) \times 80$</p> <p>$\frac{D^4 - 0.13D^4}{D} = 1527887.45$</p> <p>$\therefore D^3 = 1756192.476$</p> <p>$\therefore \boxed{D = 120.65 \text{ mm}}$</p> <p>$\therefore \boxed{d = 0.6D = 72.39 \text{ mm}}$</p>	1 1 1 1	4



Q. NO.	MODEL ANSWER	MARKS	TOTAL MARKS
6. (e) (i)	<p>Torsional eqⁿ for shaft -</p> $\frac{T}{I_p} = \frac{C\theta}{L} = \frac{\tau}{R}$ <p>Where T = Torque / Turning moment I_p or J = Polar moment of inertia of shaft Section C or G = Modulus of rigidity of shaft material θ = Angle of twist in radians L = Length of shaft f_s or τ = Maximum shear stress R = Radius of the shaft.</p>	1 1	2
6. (e) (ii)	<p>Polar modulus is the ratio of polar moment of inertia of the section and radius of the shaft</p> <p>Polar modulus $Z_p = \frac{J \text{ or } I_p}{R}$</p>	1 1	2
6. (f)	<p>Given : $D = 100 \text{ mm}$, $d = 60 \text{ mm}$, $T = 2 \text{ kN-m}$ $\omega T = 2 \times 10^3 \text{ N-m}$, $N = 300 \text{ rpm}$.</p> <p>Find Power.</p> $P = \frac{2\pi NT}{60}$ $= \frac{2 \times \pi \times 300 \times 2 \times 10^3}{60}$ <p>$P = 62,831.85 \text{ W} = 62.83 \text{ kW}$</p>	1 1 2	4