



Subject Code: 17304

SUMMER – 15 EXAMINATIONS
Model Answer- Strength of Materials

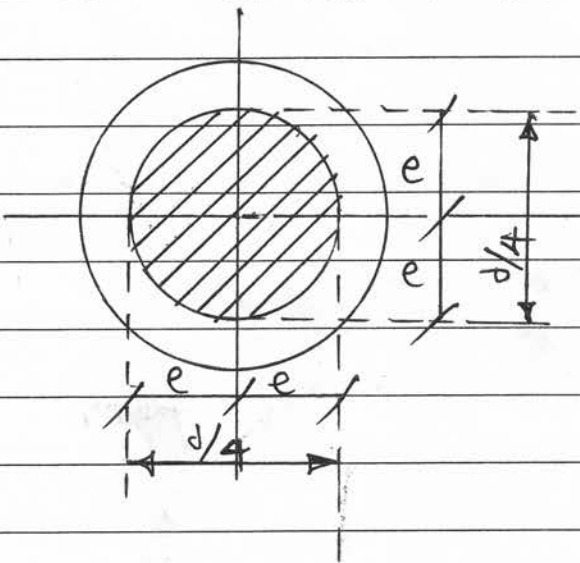
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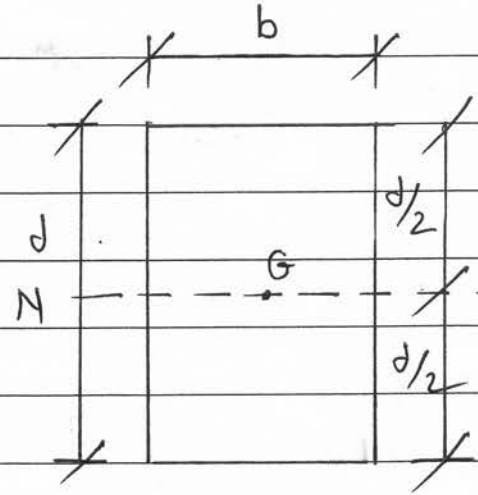
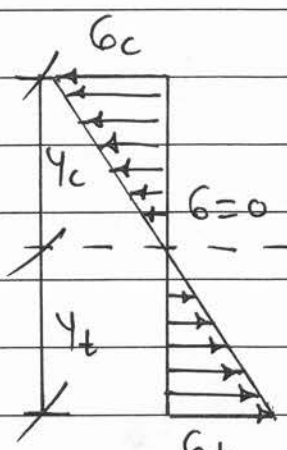
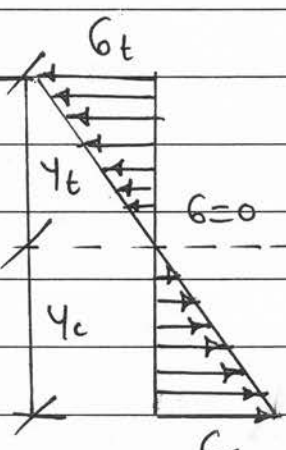
Important Instruction to Examiners:-

- 1) The answers should be examined by key words & not as word to word as given in the model answers scheme.
- 2) The model answers & answers written by the candidate may vary but the examiner may try to access the understanding level of the candidate.
- 3) The language errors such as grammatical, spelling errors should not be given more importance.
- 4) While assessing figures, examiners, may give credit for principle components indicated in the figure.
- 5) The figures drawn by candidate & model answer may vary. The examiner may give credit for any equivalent figure drawn.
- 5) Credit may be given step wise for numerical problems. In some cases, the assumed contact values may vary and there may be some difference in the candidate's answers and model answer.
- 6) In case of some questions credit may be given by judgment on part of examiner of relevant answer based on candidates understanding.
- 7) For programming language papers, credit may be given to any other programme based on equivalent concept.

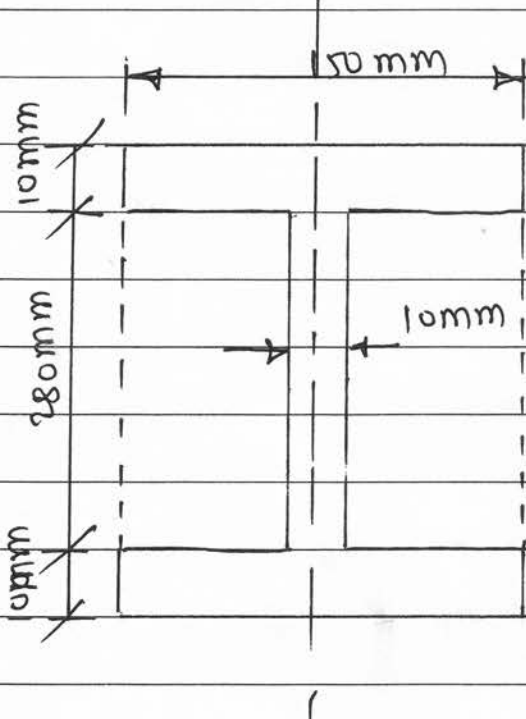
Important notes to examiner

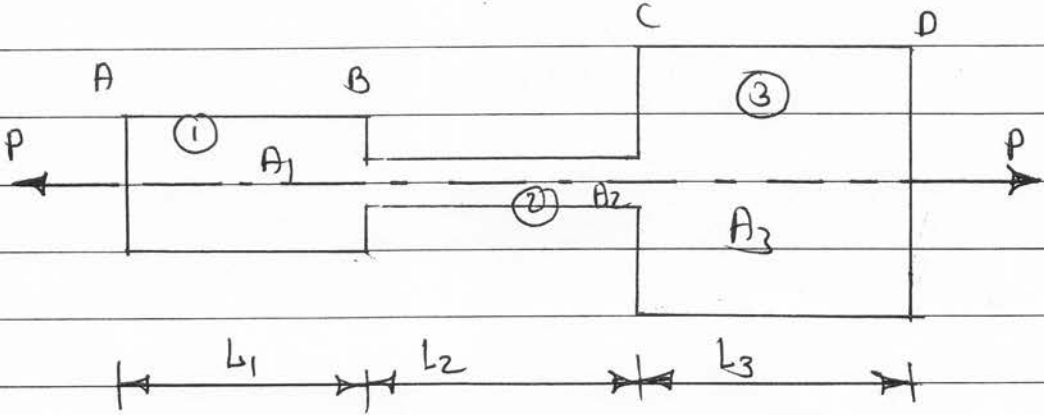
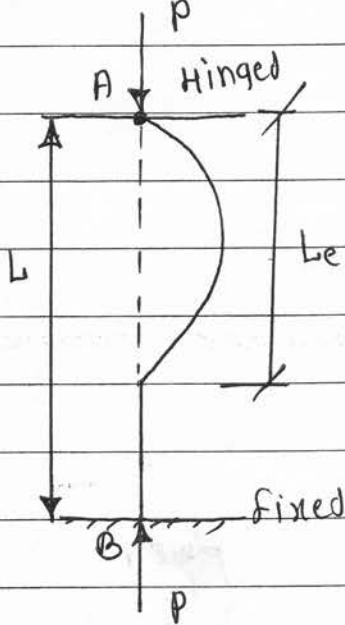
| Q. NO | SOLUTION | MARKS |
|-------|--|-------|
| Q-1 A | | |
| a) | <p>i) Fatigue —: it is property of material by virtue of it resists the failure of material under fluctuating or repeated loading</p> | 01M |
| | <p>ii) creep —: the continues deformation of material which undergoes with time due to application of external steady load is called creep.</p> | 01M |
| b) | <p>i) principal plane —: are the two planes where the normal stress (σ) is the maximum or minimum and no shear stresses.</p> | 01M |
| | <p>ii) principal stress —: The normal stresses (σ) acting on the principal planes.</p> | 01M |
| c) | $-F = \frac{\delta M}{\delta x}$ | 01M |
| | <p>the rate of change of bending moment is equal to shear force</p> | 01M |
| | <p>if $\frac{\delta F}{\delta x} = 0$, bending moment is maximum which shows the point of max. B.M and point of zero shear or the point of changing the sign of S.F</p> | |
| | | |

| Q.NO | SOLUTION | MARKS |
|------|---|--|
| (d) | <p>Assumption in theory of bending</p> <ul style="list-style-type: none"> i) The value of young's modulus of Elasticity is same in tension and compression ii) The material of beam is homogenous and isotropic. iii) The Transverse sections which were plane before bending remain plane after Bending also. iv) the Beam initially straight and all longitudinal fillament bend into circular arcs with common centre of curvature. v) The radius of curvatures large as compare to the dimensions of the cross-section vi) Each layer of the Beam is free to expand to or contract. independently of the layers, above or below it. | <p>$\frac{1}{2}$ M for <u>any</u> <u>four</u></p> |
| e) | <p>core of section for circular column.</p>  | 0.2 M |

| Q.NO | SOLUTION | MARKS |
|------|--|------------------------|
| f) | <p> $E = 3K(1-2\mu)$ where as $E = 2G(1+\mu)$ </p> <p> $E =$ young's modulus $K =$ Bulk modulus $G =$ Shear modulus $\mu =$ poisson's ratio </p> | <p>01M</p> <p>01M</p> |
| g) | <p>i) $\theta = 45^\circ$ $\theta = 135^\circ$ are angle where tangential stress is maximum</p> | <p>01M</p> <p>01M</p> |
| h) | <p>Bending stress Distribution diagram for rectangular section for simply supported & cantilever beam.</p> | |
| a) |  <p>Beam Section</p> | |
| b) |  <p>Bending stress variation diagram for simply supported beam</p> | <p>01M for (fig-b)</p> |
| c) |  <p>Bending stress variation for cantilever beam</p> | <p>01M for (fig-c)</p> |

| Q.NO | SOLUTION | MARKS |
|---------|---|-------|
| (Q-1-B) | | |
| a) | given —: | |
| | i) $P = 40 \text{ kN}$ ii) $\sigma_{\text{permissible}} = 150 \text{ MPa}$ | |
| | we have | |
| | $\sigma = \frac{P}{A} = \frac{P}{\frac{\pi}{4} d^2}$ | 01M |
| | ∴ put the value of 'P' & 'σ' | |
| | $150 = \frac{40 \times 10^3}{\frac{\pi}{4} d^2}$ | 01M |
| | $\therefore d^2 = \frac{40 \times 10^3}{\frac{\pi}{4} \times 150}$ | 01M |
| | $d^2 = \frac{40 \times 10^3}{117.81}$ | |
| | $d^2 = 339.52$ | |
| | $\boxed{d = 18.43 \text{ mm}}$ | 01M |
| b) | given data —: | |
| | i) $d = 1 \text{ m}$ ii) fluid pressure = 1.5 N/mm^2 $\sigma_{\text{tensile}} = 450 \text{ N/mm}^2$ factor of safety = 4.5 | |
| | → since hoop stress is twice the longitudinal stress, pipe will fail if the hoop stress exceeds the working stress. | |

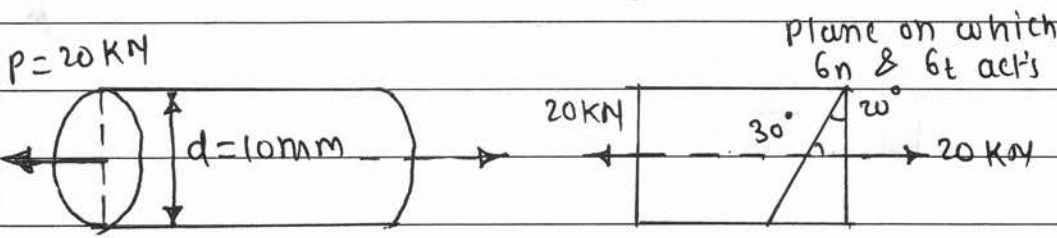
| Q. NO | SOLUTION | MARKS |
|-------|---|--------------------------------------|
| | $\therefore \text{Hoop stress } (\sigma_c) = \frac{\text{ultimate stress}}{\text{factor of safety}}$ | $\frac{1}{2} M$ |
| | $= \frac{450}{4.5} = 100 \text{ N/mm}^2$ | $\frac{1}{2} M$ |
| | $\sigma_c = \frac{pd}{2t}$ | 01 M |
| | $100 = \frac{1.5 \times 1000}{2 \times t}$ | 01 M |
| | $\therefore \text{Thickness of pipe } \boxed{t = 7.5 \text{ mm}}$ | 01 M |
| c) |  <p>given data:</p> <p>$B = 150 \text{ mm}$</p> <p>$H = 300 \text{ mm}$</p> <p>$b = \text{width of shaded portion } (150 - 10) = 140 \text{ mm}$</p> <p>$h = \text{height of shaded portion } (300 - 2 \times 10) = 280 \text{ mm}$</p> | $\frac{1}{2} M$ for <u>fig</u> |
| | $i) I_{xx} = \frac{BH^3}{12} - \frac{bh^3}{12} = \frac{1}{12} [BH^3 - bh^3]$ | 01 M |
| | $I_{xx} = \frac{1}{12} [150 \times 300^3 - 140 \times 280^3]$ | 01 M |
| | $I_{xx} = 81.39 \times 10^6 \text{ mm}^4$ | 01 M |

| Q.NO | SOLUTION | MARKS |
|--------|---|--------------------------------|
| Q-2(a) | <p>i) uniformly varying section showing axial load</p>  <p>where as $P =$ axial load A_1, A_2, A_3 & L_1, L_2, L_3 are the c/s-area & length of section ① ② ③</p> | 02M |
| ii) |  <p>effective length</p> $L_{eff} = \frac{L}{\sqrt{2}}$ | 01M for fig 01M for formula |

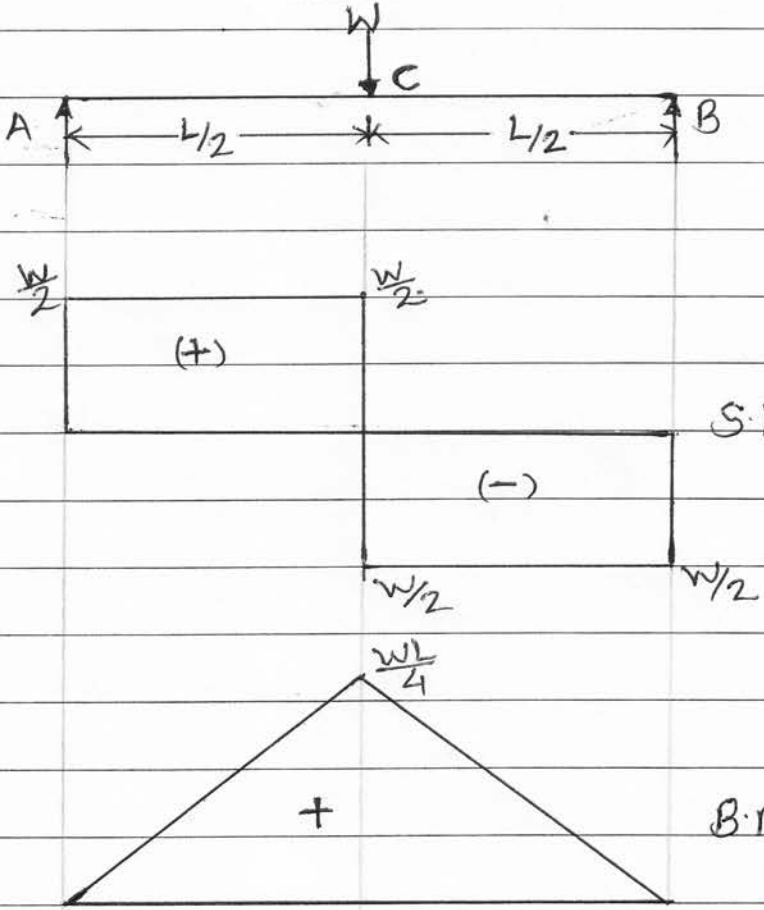
| Q.NO | SOLUTION | MARKS |
|------|--|------------------------------|
| b) | <p>Assumptions in Euler's column Theory</p> <p>i) The material of the column is perfectly homogeneous and isotropic.</p> <p>ii) The column is initially straight and of uniform lateral dimensions.</p> <p>iii) The load on the column is exactly axial.</p> <p>iv) The column is long and fails due to buckling or bending only.</p> <p>v) The self weight of the column is neglected.</p> <p>vi) The column is stressed upto the limit of proportionality.</p> | 0.1 M (For each any four) |
| c) | <p>given data</p> <p>$L = 300 \text{ mm}$ $d = 20 \text{ mm}$</p> <p>$t = 100^\circ \text{C}$ Total extension = 0.4 mm</p> <p>$E = 2 \times 10^5 \text{ N/mm}^2$ $\alpha = 12 \times 10^{-6} / ^\circ \text{C}$</p> | $\frac{1}{2} \text{ M}$ |
| | <p>Total extension of the rod = free expansion + Extension δL due to force 'P'</p> | |
| | <p>$0.4 = (L \alpha t) + \delta L$ due to force 'P'</p> | $\frac{1}{2} \text{ M}$ |
| | <p>$0.4 = (300 \times 12 \times 10^{-6} \times 100) + \delta L$</p> | |
| | <p>$0.4 = 0.36 + \delta L$</p> | |
| | <p>$\therefore \delta L = 0.04 \text{ mm}$</p> | $\frac{1}{2} \text{ M}$ |

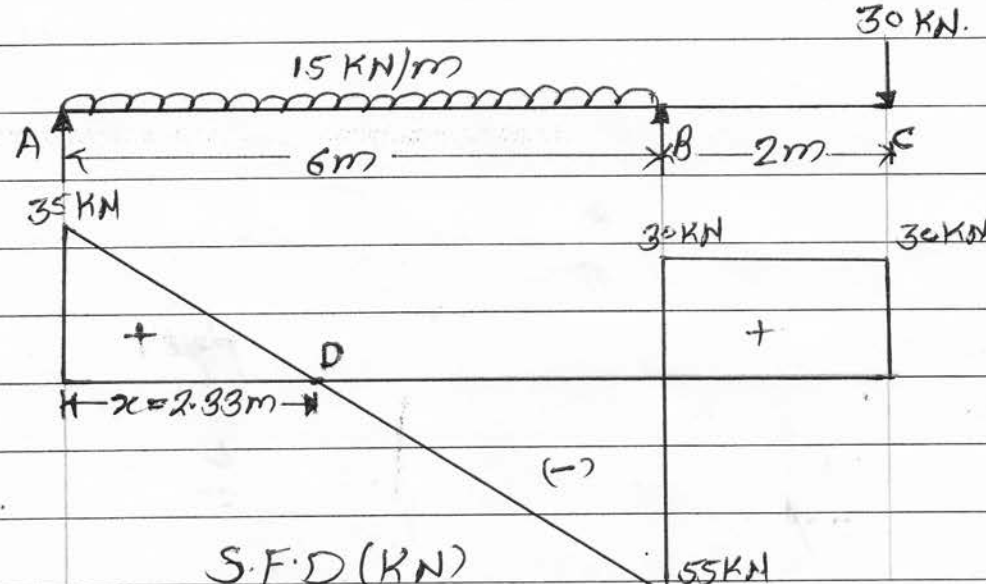
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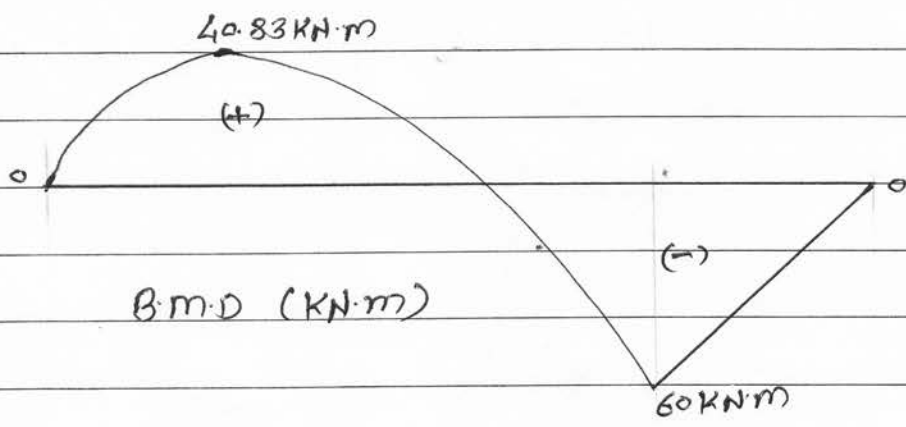
| Q. NO | SOLUTION | MARKS |
|-------------|---|------------------|
| Q-2 (c) | But | |
| (continued) | $\delta L = \frac{P \cdot L}{AE}$ | $\frac{1}{2} M$ |
| | $0.04 = \frac{P \times 300}{\left(\frac{\pi \times 20^2}{4}\right) \times 2 \times 10^5}$ | $\frac{1}{2} M$ |
| | $\therefore P = \frac{0.04 \times \left(\frac{\pi}{4} \times 20^2\right) \times 2 \times 10^5}{300}$ | $\frac{1}{2} M$ |
| | $\therefore P = 8.377 \times 10^3 \text{ N}$ <u>or</u> $P = 8.377 \text{ kN}$ | 01 M |
| d) | | |
| | <p>i) let $A_1 = 40 \times 40 = 1600 \text{ mm}^2$ $L_1 = 1 \text{ m} = 1000 \text{ mm}$ $A_2 = 20 \times 20 = 400 \text{ mm}^2$ $L_2 = 1 \text{ m} = 1000 \text{ mm}$ $A_3 = 30 \times 30 = 900 \text{ mm}^2$ $L_3 = 1.5 \text{ m} = 1500 \text{ mm}$</p> | |
| | <p>ii) To find unknown value of 'P' $\sum F_x = -120 + 220 - P + 160$ $P = 260 \text{ kN}$ (\leftarrow)</p> | 01 M |
| | <p>iii) $\delta L = \delta L_1 + \delta L_2 + \delta L_3$ $\delta L = \frac{P_1 L_1}{A_1 E} + \frac{P_2 L_2}{A_2 E} + \frac{P_3 L_3}{A_3 E}$</p> | $\frac{01 M}{2}$ |

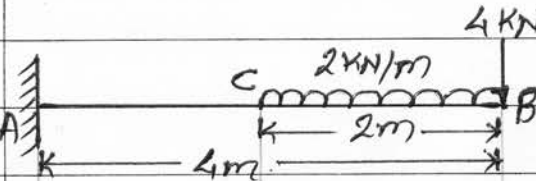
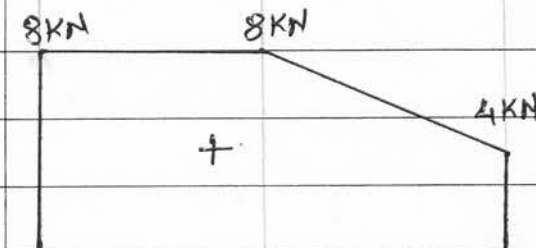
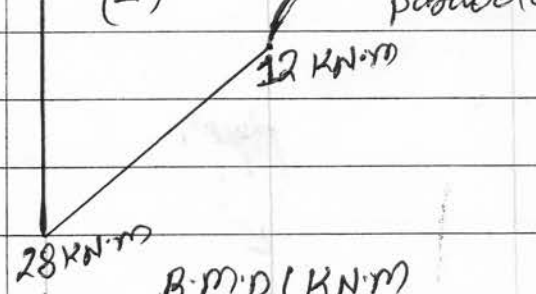
| Q.NO | SOLUTION | MARKS |
|------|--|-------|
| | $\delta L = \frac{1}{E} \left[\frac{P_1 L_1}{A_1} + \frac{P_2 L_2}{A_2} + \frac{P_3 L_3}{A_3} \right]$ | 1/2 M |
| | $= \frac{1}{2 \times 10^5} \left[\frac{(120 \times 10^3) \times 1000}{1600} + \frac{(-100 \times 10^3) \times 1000}{400} \right.$ | |
| | $\left. + \frac{(160 \times 10^3) \times 1000}{900} \right]$ | 0.1 M |
| | $= \frac{1}{2 \times 10^5} [75 \times 10^3 - 250 \times 10^3 + 266.67 \times 10^3]$ | |
| | $\delta L = 0.458 \text{ mm}$ | 0.1 M |
| e) | given data | |
| | d = 10 mm | |
| | P _x = axial pull = 20 kN = 20 × 10 ³ N, q _y = 0, q _z = 0 | |
| |  | 1/2 M |
| | $i) A = \frac{\pi}{4} \times d^2 = \frac{\pi}{4} \times 10^2 = 78.539 \text{ mm}^2$ | 1/2 M |
| | $ii) \sigma_x = \frac{P_x}{A} = \frac{20 \times 10^3}{78.539} = 254.65 \text{ N/mm}^2$ | 1/2 M |
| | $\theta = 90^\circ - 30^\circ = 60^\circ$ | |

| Q.NO | SOLUTION | MARKS |
|------|---|-------|
| | iii) To find normal stress (σ_n) and tangential stress (σ_t) | |
| | $\sigma_n = \left(\frac{\sigma_x + \sigma_y}{2} \right) + \left(\frac{\sigma_x - \sigma_y}{2} \right) \cos 2\theta + q \cdot \sin 2\theta$ $= \left(\frac{254.65 + 0}{2} \right) + \left(\frac{254.65 - 0}{2} \right) \cos(2 \times 60^\circ) + 0$ | 1/2 M |
| | $\sigma_n = \frac{254.65}{2} + \frac{254.65 - 0}{2} \cos 120^\circ$ | |
| | $\sigma_n = 63.66 \text{ N/mm}^2 \text{ (Tensile)}$ | 1/2 M |
| | $\sigma_t = \left(\frac{\sigma_x - \sigma_y}{2} \right) \sin 2\theta - q \cdot \cos 2\theta$ | 1/2 M |
| | $= \left(\frac{254.65 - 0}{2} \right) \sin(2 \times 60^\circ) - 0$ | 1/2 M |
| | $\sigma_t = \frac{254.65}{2} \sin 120^\circ$ | |
| | $\sigma_t = 110.266 \text{ N/mm}^2$ | 1/2 M |
| | <p>f) given data : $L = 3\text{m} = 3000\text{mm}$ $d = 1\text{m} = 100\text{mm}$ $t = 15\text{mm}$, $p = 1.5 \text{ N/mm}^2$ $E = 2 \times 10^5 \text{ N/mm}^2$ $\mu = 0.25$</p> | 1/2 M |
| | <p>i) To find circumferential strain (e_c)</p> $e_c = \frac{pd}{4tE} (2 - \mu)$ | 1/2 M |

| Q.NO | SOLUTION | MARKS |
|-------|--|---|
| Q3 a) |  <p style="text-align: center;">W ↓ C</p> <p style="text-align: center;">A ← L/2 * L/2 → B</p> <p style="text-align: center;">W/2 W/2</p> <p style="text-align: center;">(+)</p> <p style="text-align: center;">S.F.D.</p> <p style="text-align: center;">(-)</p> <p style="text-align: center;">W/2 W/2</p> <p style="text-align: center;">WL/4</p> <p style="text-align: center;">+</p> <p style="text-align: center;">B.M.D.</p> | <p style="text-align: right;">$\frac{1}{2}m$</p> <p style="text-align: right;">$\frac{1}{2}m$</p> |
| i) | Support reactions | |
| a) | $\sum F_y = 0$ | |
| | $R_A + R_B = W \quad \text{--- (i)}$ | |
| b) | $\sum M @ A = 0$ | |
| | $W \times \frac{L}{2} - R_B \times L = 0$ | |
| | $\therefore R_B = \frac{WL}{2 \cdot L} = \frac{W}{2}$ | $\frac{1}{2}m$ |
| | $\therefore R_A = W - R_B = W - \frac{W}{2} = \frac{W}{2}$ | $\frac{1}{2}m$ |
| ii) | S.F. Calculation | |
| | S.F. at just left of A = 0 KN | |
| | S.F. at just right of A = $R_A = \frac{W}{2}$ KN | |
| | S.F. at just left of C = $\frac{W}{2}$ KN | |

| Q.NO | SOLUTION | MARKS |
|----------|--|----------------|
| Q3a) | S.F at just right of C = $\frac{W}{2} - W = -\frac{W}{2}$ KN | |
| Conto... | S.F at just left of B = $-\frac{W}{2}$ KN | 1M |
| | S.F at just right of B = $-\frac{W}{2} + R_B = 0$ KN | |
| | iii) B.M calculation | |
| | B.M at A = B.M at B = 0 KN.m ... simple support | |
| | * As the point of Contraflexure is under the point load the maximum B.M will be developed under the point load i.e. at C | |
| | ∴ Max. B.M _C = $R_A \times \frac{L}{2} = \frac{W}{2} \times \frac{L}{2} = \frac{WL}{4}$ | $\frac{1}{2}M$ |
| | * The maximum shear force will be developed at the supports, due to symmetry maximum S.F at support A & B is equal | |
| | Maximum S.F = R_A or $R_B = \frac{W}{2}$ | $\frac{1}{2}M$ |
| Q3b) |  <p>The diagram shows a beam with a uniformly distributed load (UDL) of 15 kN/m over a length of 6m from support A to support B. A point load of 30 kN is applied at point F, which is 2m to the right of support B. The reaction force at support A is 35 kN, and at support B is 30 kN. The shear force diagram (S.F.D) is plotted below the beam. It starts at 35 kN at support A, decreases linearly to -55 kN at support B, crossing the zero line at point D, which is 2.93m from support A. Between support B and point F, the shear force is constant at 30 kN. The area under the S.F.D is marked with '+' for positive and '-' for negative.</p> | $\frac{1}{2}M$ |

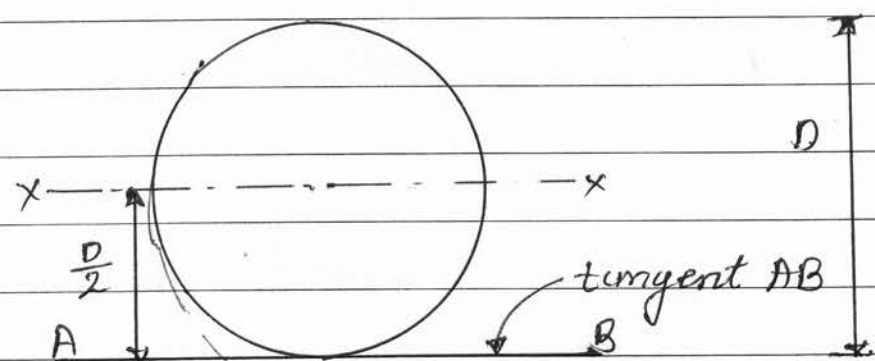
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|---------|--|-----------------|
| Q3b) | | |
| Cont... |  <p style="text-align: center;">B.M.D (KN.m)</p> | $\frac{1}{2}$ m |
| | <p>i) Support reaction</p> <p>a) $\sum F_y = 0$</p> $R_A + R_B = (15 \times 6) + 30 = 120 \text{ KN} \quad \text{--- (i)}$ <p>b) $\sum M @ A = 0$</p> $(15 \times 6 \times 3) + (30 \times 8) - R_B \times 6 = 0$ $510 = 6 R_B$ | |
| | $\therefore R_B = 85 \text{ KN}$ $\therefore R_A = 35 \text{ KN}$ | 1 m |
| | <p>ii) S.F. Calculation</p> <p>S.F at just left of A = 0 KN</p> <p>S.F at just right of A = $R_A = 35 \text{ KN}$</p> <p>S.F at just left of B = $35 - (15 \times 6) = -55 \text{ KN}$</p> <p>S.F at just right of B = $-55 + 85 = 30 \text{ KN}$</p> <p>S.F at just left of C = 30 KN</p> <p>S.F at just right of C = $30 - 30 = 0 \text{ KN}$.</p> | 1 m |
| | <p>Let D be the point of contraflexure.</p> <p>from similar triangles</p> $\frac{35}{x} = \frac{55}{6-x}$ | |

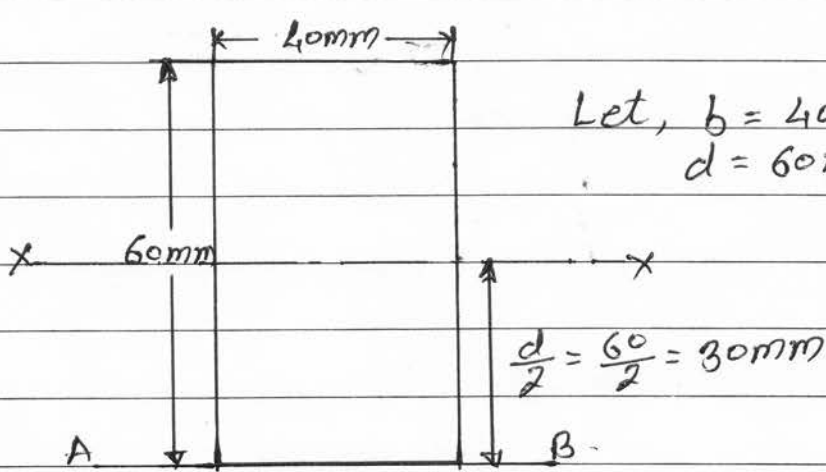
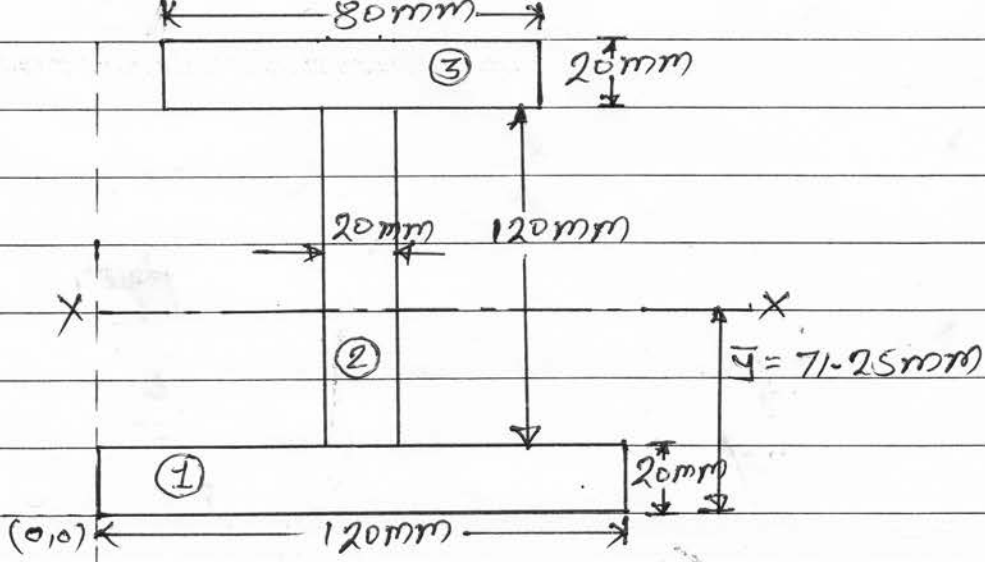
| Q.NO | SOLUTION | MARKS |
|---------|---|------------------|
| Q 3b | | |
| Cont... | $\therefore x = \frac{35 \times 6}{35 + 55} = 2.33 \text{ m from A.}$ | |
| | <p>iii) B.M. Calculation</p> | |
| | <p>B.M at A = 0 simple support</p> | |
| | <p>B.M at C = 0 free end of overhang.</p> | 1M |
| | <p>B.M at D = $RA \times 2.33 - (15 \times 2.33 \times \frac{2.33}{2})$</p> | |
| | <p>B.M at D = 40.83 KN.m</p> | |
| | <p>B.M at B = $RA \times 6 - (15 \times 6 \times 3) = -60 \text{ KN.m.}$</p> | |
| | | |
| | | |
| | | |
| Q 3c) |  | 1M for SFD & BMD |
| | <p>i) Support reactions</p> | |
| | <p>$\sum F_y = 0$; $RA = 4 + (2 \times 2) = 8 \text{ KN}$</p> | 1M |
| | | |
| |  | |
| | <p>ii) S.F. calculation</p> | |
| | <p>S.F at just left of A = 0 KN</p> | |
| | <p>S.F at just right of A = $RA = 8 \text{ KN}$</p> | |
| | <p>S.F at just C = 8 KN</p> | 1M |
| | <p>S.F at just left of B = $8 - (2 \times 2) = 4 \text{ KN}$</p> | |
| | <p>S.F at just right of B = $4 - 4 = 0 \text{ KN.}$</p> | |
| | | |
| |  | |
| | <p>iii) B.M. calculation.</p> | |
| | <p>B.M at B = 0 ... free end.</p> | |
| | <p>B.M at C = $-4 \times 2 - 2 \times 2 \times 1$</p> | 1M |
| | <p>= -12 KN.m</p> | |
| | <p>B.M at A = $-4 \times 4 - 2 \times 2 \times 3$</p> | |
| | <p>= -28 KN.m</p> | |

| Q.NO | SOLUTION | MARKS |
|-----------------------|--|---|
| Q3 d) | <p style="text-align: center;">S.F.D (KN)</p> <p style="text-align: center;">B.M.D (KN.m)</p> | <p style="text-align: right;">$\frac{1}{2}$ m</p> <p style="text-align: right;">$\frac{1}{2}$ m</p> |
| i) Support reaction | $\sum F_y = 0 ; R_A = 2 + (1 \times 2) = 4 \text{ kN.}$ | 1 m |
| ii) S.F. calculation | <p>S.F. at left of A = 0 kN</p> <p>S.F. at right of A = $R_A = 4 \text{ kN}$</p> <p>S.F. at left of c = 4 kN.</p> <p>S.F. at right of c = $4 - 2 = 2 \text{ kN}$</p> <p>S.F. at D = 2 kN.</p> <p>S.F. at B = $2 - (1 \times 2) = 0 \text{ kN.}$</p> | 1 m |
| iii) B.M. calculation | <p>B.M. at B = 0 free end.</p> <p>B.M. at D = $-(1 \times 2 \times 1) = -2 \text{ kNm}$</p> <p>B.M. at c = $-(1 \times 2 \times 2) = -4 \text{ kNm}$</p> <p>B.M. at A = $-(1 \times 2 \times 3) - (2 \times 1) = -8 \text{ kNm}$</p> | 1 m |

| Q.NO | SOLUTION | MARKS |
|------|--|---|
| Q3e) | <p style="text-align: center;">S.F.D (KN)</p> <p style="text-align: center;">B.M.D (KN.M)</p> | <p style="text-align: center;">$\frac{1}{2}$ M</p> <p style="text-align: center;">$\frac{1}{2}$ M</p> |
| | <p>i) Support reactions</p> <p>or $\sum F_y = 0$</p> <p>$R_A + R_B = 5 + 7 = 12 \text{ KN}$.</p> <p>b) $\sum M @ A = 0$</p> <p>$(5 \times 1.5) + (7 \times 3.5) - 4R_B = 0$</p> <p>$32 = 4R_B$</p> <p>$\therefore R_B = 8 \text{ KN}$</p> <p>$\therefore R_A = 12 - 8 = 4 \text{ KN}$.</p> <p>ii) S.F. Calculation</p> <p>S.F at just left of A = 0</p> <p>S.F at just right of A = $R_A = 4 \text{ KN}$</p> <p>S.F at just left of C = 4 KN</p> <p>S.F at just right of C = $4 - 5 = -1 \text{ KN}$.</p> | <p style="text-align: center;">1 M</p> |

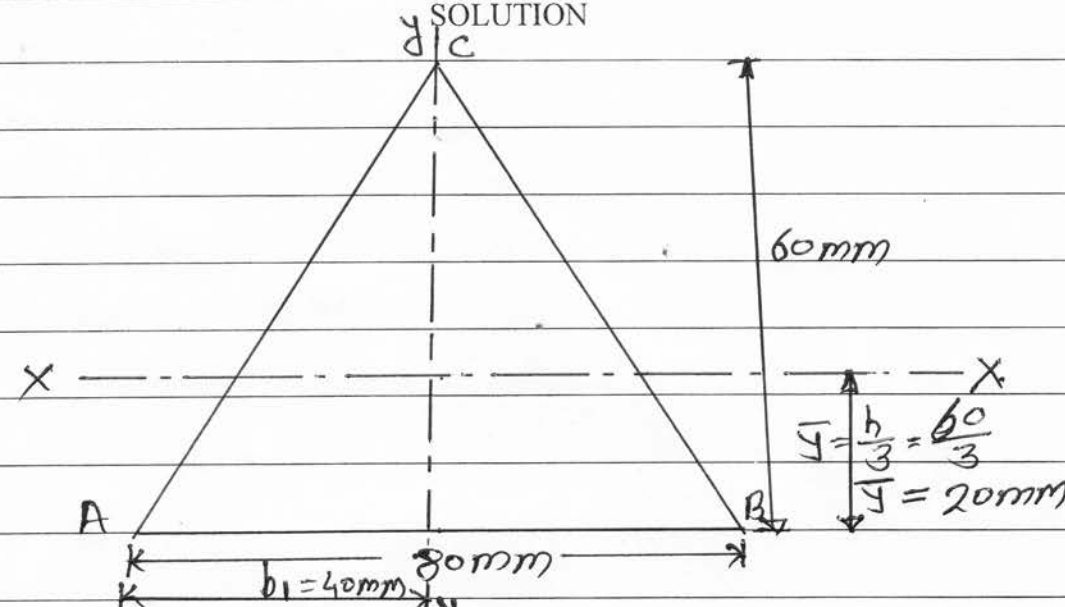
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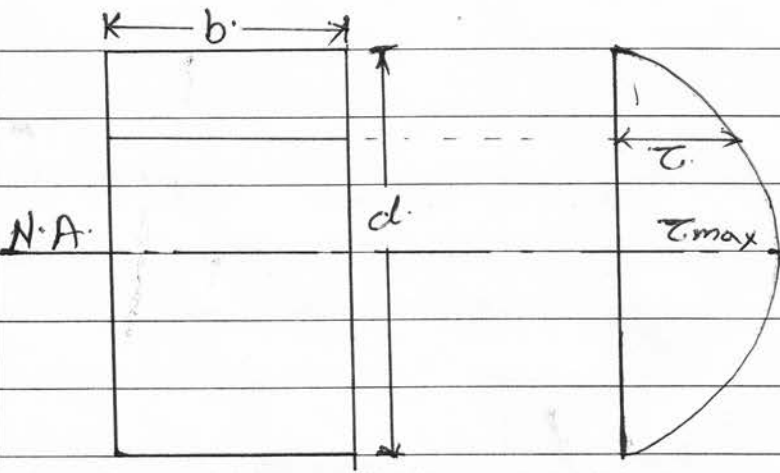
| Q.NO | SOLUTION | MARKS |
|----------|--|-------|
| Q3e) | S.F. at just left of D = -1 kN/m | |
| Conto... | S.F. at just right of D = -1 - 7 = -8 kN | 1M |
| | S.F. at just left of B = -8 kN | |
| | S.F. at just right of B = -8 + R _B = 0 kN. | |
| | iii) B.M. Calculation. | |
| | B.M. at A = B.M. at B = 0 ... simple support | |
| | B.M. at C = R _A × 1.5 = 4 × 1.5 = 6 kN/m | 1M |
| | B.M. at D = R _A × 3.5 - (5 × 2) = 4 × 3.5 - 10 = 4 kN/m. | |
| Q3f) |  | 1M |
| | Given - $M I_{AB} = 6.283 \times 10^5 \text{ mm}^4$ — (i) | |
| | Using parallel axis theorem $M I_{AB}$ is given by | |
| | $M I_{AB} = I_G + A h^2 = \left[\frac{\pi D^4}{64} + \frac{\pi D^2}{4} \times \left(\frac{D}{2}\right)^2 \right]$ | 1M |
| | $M I_{AB} = \frac{\pi D^4}{64} + \frac{\pi D^4}{16} = 0.2454 D^4$ — (ii) | 1M |
| | Equating eq ⁿ (i) & (ii) | |
| | $6.283 \times 10^5 = 0.2454 D^4$ | |
| | $D^4 = 2.56 \times 10^6$ | |
| | $D = (2.56 \times 10^6)^{1/4}$ | |
| | $D = 40 \text{ mm}$ | 1M |
| | Diameter of disc is 40 mm. | |

| Q.NO | SOLUTION | MARKS |
|-------|---|-------|
| Q4 a) |  <p>Let, $b = 40\text{mm}$ $d = 60\text{mm}$</p> <p>$\frac{d}{2} = \frac{60}{2} = 30\text{mm}$</p> | 1M |
| | <p>M.I. about AB</p> <p>By parallel axis theorem.</p> $I_{AB} = I_G + Ah^2$ $= \frac{bd^3}{12} + bd \cdot \left(\frac{d}{2}\right)^2$ | 1M |
| | $= \frac{40 \times 60^3}{12} + 40 \times 60 \times \left(\frac{60}{2}\right)^2$ | 1M |
| | $= 720 \times 10^3 + 2.16 \times 10^6$ $I_{AB} = 2.88 \times 10^6 \text{ mm}^4$ | 1M |
| | <p>M.I. of solid rectangular section about its smaller side is $2.88 \times 10^6 \text{ mm}^4$</p> | |
| Q4 b) |  <p>$\bar{y} = 71.25\text{mm}$</p> | |

| Q.NO | SOLUTION | MARKS | | | | | | | | | |
|--|---|--|--|---|-----------------------|-----------------------|-----------------------|--------------------------------------|--|--|--|
| Q4b) | i) Position of X-X axis (\bar{Y}) | | | | | | | | | | |
| Cont... | <table border="1" style="width: 100%; border-collapse: collapse;"> <tr> <td style="width: 33%;">$a_1 = 120 \times 20$ $= 2400 \text{ mm}^2$</td> <td style="width: 33%;">$a_2 = 120 \times 20$ $= 2400 \text{ mm}^2$</td> <td style="width: 33%;">$a_3 = 80 \times 20$ $= 1600 \text{ mm}^2$</td> </tr> <tr> <td>$x_1 = 60 \text{ mm}$</td> <td>$x_2 = 60 \text{ mm}$</td> <td>$x_3 = 60 \text{ mm}$</td> </tr> <tr> <td>$y_1 = \frac{20}{2} = 10 \text{ mm}$</td> <td>$y_2 = 20 + \frac{120}{2} = 80 \text{ mm}$</td> <td>$y_3 = 20 + 120 + \frac{20}{2} = 150 \text{ mm}$</td> </tr> </table> | $a_1 = 120 \times 20$ $= 2400 \text{ mm}^2$ | $a_2 = 120 \times 20$ $= 2400 \text{ mm}^2$ | $a_3 = 80 \times 20$ $= 1600 \text{ mm}^2$ | $x_1 = 60 \text{ mm}$ | $x_2 = 60 \text{ mm}$ | $x_3 = 60 \text{ mm}$ | $y_1 = \frac{20}{2} = 10 \text{ mm}$ | $y_2 = 20 + \frac{120}{2} = 80 \text{ mm}$ | $y_3 = 20 + 120 + \frac{20}{2} = 150 \text{ mm}$ | |
| $a_1 = 120 \times 20$ $= 2400 \text{ mm}^2$ | $a_2 = 120 \times 20$ $= 2400 \text{ mm}^2$ | $a_3 = 80 \times 20$ $= 1600 \text{ mm}^2$ | | | | | | | | | |
| $x_1 = 60 \text{ mm}$ | $x_2 = 60 \text{ mm}$ | $x_3 = 60 \text{ mm}$ | | | | | | | | | |
| $y_1 = \frac{20}{2} = 10 \text{ mm}$ | $y_2 = 20 + \frac{120}{2} = 80 \text{ mm}$ | $y_3 = 20 + 120 + \frac{20}{2} = 150 \text{ mm}$ | | | | | | | | | |
| | $\therefore \bar{Y} = \frac{a_1 y_1 + a_2 y_2 + a_3 y_3}{a_1 + a_2 + a_3}$ $= \frac{(2400 \times 10) + (2400 \times 80) + (1600 \times 150)}{2400 + 2400 + 1600}$ | | | | | | | | | | |
| | $\bar{Y} = 71.25 \text{ mm}$ from bottom. | 1m | | | | | | | | | |
| | ii) M.I. about X-X by parallel axis theorem | | | | | | | | | | |
| | $I_{XX} = I_{XX1} + I_{XX2} + I_{XX3}$ $I_{XX} = [I_{G1} + A_1 h_1^2] + [I_{G2} + A_2 h_2^2] + [I_{G3} + A_3 h_3^2]$ | 1m | | | | | | | | | |
| | $I_{XX1} = \frac{120 \times 20^3}{12} + 2400(71.25 - 10)^2 = 9.08 \times 10^6 \text{ mm}^4$ $\frac{1}{2} \text{ m}$ | | | | | | | | | | |
| | $I_{XX2} = \frac{20 \times 120^3}{12} + 2400(71.25 - 80)^2 = 3.06 \times 10^6 \text{ mm}^4$ $\frac{1}{2} \text{ m}$ | | | | | | | | | | |
| | $I_{XX3} = \frac{80 \times 20^3}{12} + 1600(71.25 - 150)^2 = 9.97 \times 10^6 \text{ mm}^4$ $\frac{1}{2} \text{ m}$ | | | | | | | | | | |
| | $\therefore I_{XX} = 9.08 \times 10^6 + 3.06 \times 10^6 + 9.97 \times 10^6$ $I_{XX} = 22.11 \times 10^6 \text{ mm}^4$ | | | | | | | | | | |
| | \therefore M.I. about X-X axis is $22.11 \times 10^6 \text{ mm}^4$ | $\frac{1}{2} \text{ m}$ | | | | | | | | | |

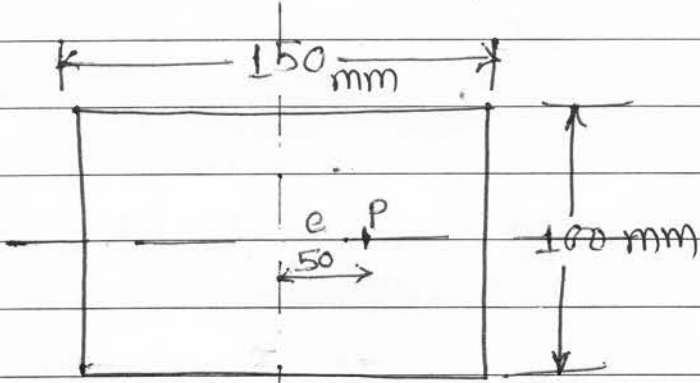
| Q.NO | SOLUTION | MARKS |
|------|--|-------------------------|
| Q4C) | | |
| | <p>i) Position of y-y axis (\bar{x})</p> $a_1 = 115 \times 10 = 1150 \text{ mm}^2 \quad a_2 = 75 \times 10 = 750 \text{ mm}^2$ $x_1 = \frac{10}{2} = 5 \text{ mm} \quad x_2 = \frac{75}{2} = 37.5 \text{ mm}$ | |
| | $\bar{x} = \frac{a_1 x_1 + a_2 x_2}{a_1 + a_2} = \frac{(1150 \times 5) + (750 \times 37.5)}{1150 + 750}$ | |
| | $\bar{x} = 17.828 \text{ mm}$ | 1M |
| | <p>ii) M.I. about y-y is given by parallel axis theorem</p> $I_{yy} = I_{yy_1} + I_{yy_2}$ | 1M |
| | $I_{yy_1} = I_{G_1} + A_1 h_1^2 = \left[\frac{db^3}{12} + bd(\bar{x} - x_1)^2 \right] \quad \text{--- (1)}$ | |
| | $I_{yy_1} = \frac{115 \times 10^3}{12} + 1150 (17.828 - 5)^2 = 198.82 \times 10^3 \text{ mm}^4 \quad \frac{1}{2} \text{ M}$ | |
| | $I_{yy_2} = I_{G_2} + A_2 h_2^2 = \frac{10 \times 75^3}{12} + 750 (17.828 - 37.5)^2$ | |
| | $I_{yy_2} = 641.80 \times 10^3 \text{ mm}^4$ | $\frac{1}{2} \text{ M}$ |
| | $\therefore I_{yy} = 840.62 \times 10^3 \text{ mm}^4$ | 1M |

| Q.NO | SOLUTION | MARKS |
|------|--|----------------|
| Q4d) |  | |
| i) | <p>M.I. about the G.G of section</p> $I_{xx} = \frac{bh^3}{36}$ | $\frac{1}{2}M$ |
| | $I_{xx} = \frac{80 \times 60^3}{36} = 480 \times 10^3 \text{ mm}^4$ | $\frac{1}{2}M$ |
| | $I_{yy} = \frac{hb^3}{12} \times 2 = \frac{2 \times 60 \times 40^3}{12} = 640 \times 10^3 \text{ mm}^4$ | $1M$ |
| ii) | <p>M.I. about the base of triangle.</p> <p>I_{AB} = By parallel axis theorem</p> $= I_G + Ah^2 = \frac{bh^3}{36} + \frac{1}{2} \times b \times h \times \left(\frac{h}{3}\right)^2$ | |
| | $I_{AB} = \frac{bh^3}{36} + \frac{bh^3}{18} = \frac{bh^3}{12}$ | $1M$ |
| | $\therefore I_{AB} = \frac{80 \times 60^3}{12} = 1.44 \times 10^6 \text{ mm}^4$ | $1M$ |
| | <p><u>OR</u> <u>alternate solution</u></p> | |
| iii) | $I_{AB} = \frac{bh^3}{12} = \frac{80 \times 60^3}{12} = 1.44 \times 10^6 \text{ mm}^4$ | $2M$ |

| Q.NO | SOLUTION | MARKS |
|-------|---|-------|
| Q4 e) | i) Bending eq ⁿ . | |
| | $\frac{M}{I} = \frac{\sigma}{y} = \frac{E}{R}$ | 2M |
| | ii) Moment of resistance (M.R) The resistance offered by the moment of couple due to two equal & opposite forces to the bending moment m due to external loads is called as moment of resistance. moment of resistance is the internal moment setup in the material of the beam. | 2M |
| Q4 f) |  <p>Rectangular section Shear stress distribution diagram.</p> | 2M |
| | i) Relationship between maximum & avg. shear stress | |
| | $\tau_{max} = \frac{3}{2} \tau_{avg}$ | 2M |
| | $\tau_{max} = 1.5 \tau_{avg}$ | |

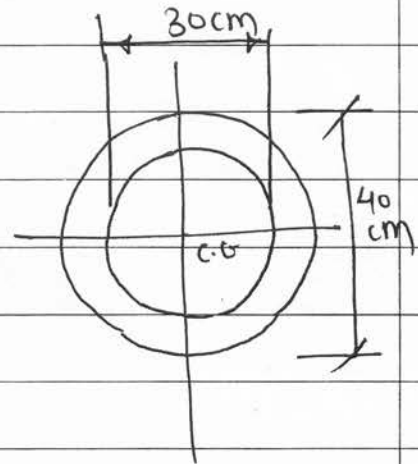
| Q.NO | SOLUTION | MARKS |
|------|---|-------------------------|
| 5-d) | Given:- for rectangular beam | |
| | $b = 100 \text{ mm}$, $d = 150 \text{ mm}$ | |
| | $l = 2 \text{ m}$, $\sigma_b = \text{Maximum bending stress} = 28 \text{ N/mm}^2$ | |
| | $\tau_{\text{max}} = \text{shear stress} = 2 \text{ N/mm}^2$ | |
| | To find:- Intensity of 'udl' i.e. Maximum load supported by beam in N/m . | |
| | Sol ⁿ :- $\frac{M}{I} = \frac{\sigma_b}{y}$ | $\frac{1}{2} \text{ M}$ |
| | $M = \text{Maximum bending moment}$ | |
| | $= \frac{wl^2}{8} \rightarrow \text{Assuming beam as simply supported beam}$ | |
| | $M = \frac{w \times (2)^2}{8} = 0.5w \text{ Nm}$ | |
| | $M = 0.5w \times 10^3 \text{ N-mm}$ | $\frac{1}{2} \text{ M}$ |
| | $I = \frac{bd^3}{12} = \frac{100 \times (150)^3}{12} = 28.125 \times 10^6 \text{ mm}^4$ | $\frac{1}{2} \text{ M}$ |
| | $y = \frac{d}{2} = \frac{150}{2} = 75 \text{ mm}$. | $\frac{1}{2} \text{ M}$ |
| | $\therefore \frac{M}{I} = \frac{\sigma_b}{y}$ | |
| | $\frac{0.5w \times 10^3}{28.125 \times 10^6} = \frac{28}{75}$ | 0.1 M |
| | $w = \frac{28 \times 28.125 \times 10^6}{0.5 \times 10^3 \times 75}$ | |
| | $w = 21000 \text{ N/m}$ | |
| | $w = 21 \text{ kN/m} \rightarrow \text{Ans. (udl)}$ | |
| | $w = 21 \times 2 = 42 \text{ kN} \rightarrow \text{Point load}$ | 0.1 M |

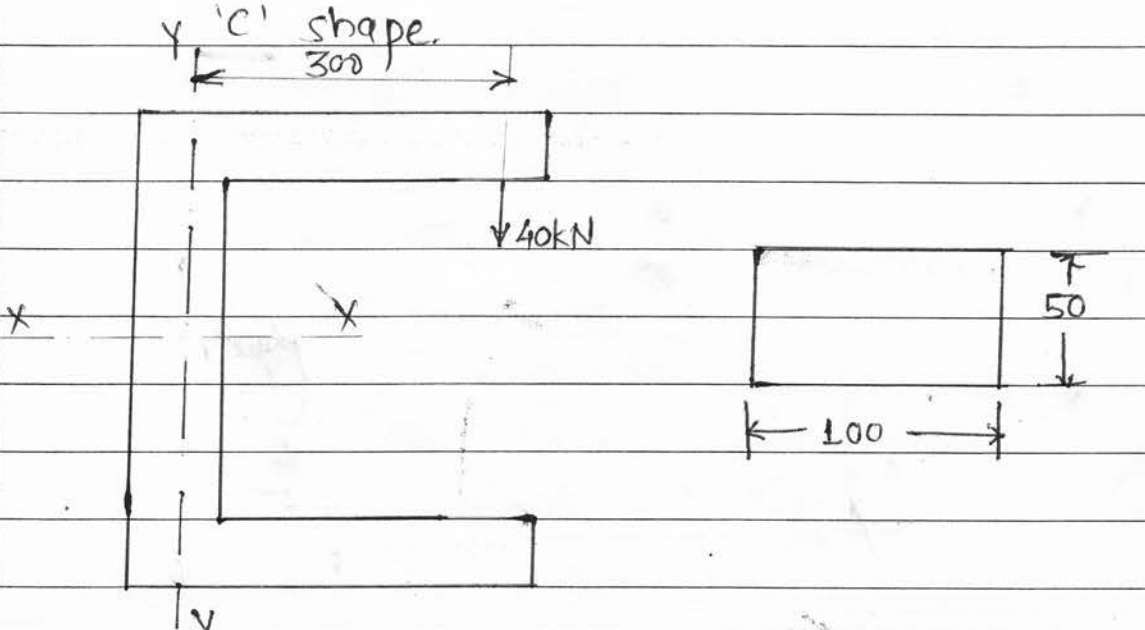
| Q.NO | SOLUTION | MARKS |
|------|---|-------|
| 5-b | Given:- d = diameter of shaft i.e. circular section d = 80mm. | |
| | To find:- limit of eccentricity 'e' | |
| | <u>solution:-</u> We know that | |
| | $\sigma_o = \sigma_b$ | 1/2 M |
| | $\frac{P}{A} = \frac{M}{Z}$ | 1/2 M |
| | $\frac{P}{A} = \frac{Pe}{Z}$ | |
| | $\frac{1}{A} = \frac{e}{Z}$ | |
| | $e = \frac{Z}{A}$ | 1/2 M |
| | $e = \frac{\frac{\pi}{32} \times d^3}{\frac{\pi}{4} \times d^2}$ | 1/2 M |
| | $= \frac{\pi}{32} \times \frac{d^3}{1} \times \frac{4}{\pi} \times \frac{1}{d^2}$ | |
| | $e = \frac{d}{8}$ | 0.1 M |
| | $e = \frac{80}{8}$ | |
| | $e = 10\text{mm} \rightarrow \text{Ans.}$ | 0.1 M |

| Q.NO | SOLUTION | MARKS |
|------|--|-------------------------|
| 5-C | <u>Given:-</u> for a rectangular column | |
| |  <p style="text-align: center;"> $b = \text{width of rectangular plate} = 150 \text{ mm}$ $d = \text{thickness of column} = 100 \text{ mm}$ $P = 150 \times 10^3 \text{ N, load}$ $e = \text{eccentricity} = 50 \text{ mm}$ </p> | |
| | <p>To find: (i) Maximum stress intensity (σ_{\max}) (ii) Minimum stress intensity (σ_{\min})</p> <p><u>Solⁿ</u>:- We know that,</p> $\sigma_{\max} = \sigma_0 + \sigma_b$ $\sigma_{\min} = \sigma_0 - \sigma_b$ $\sigma_0 = \frac{P}{A} = \frac{150 \times 10^3}{100 \times 150}$ $= \frac{150 \times 10^3}{150 \times 100}$ | |
| | $\sigma_0 = 10 \text{ N/mm}^2$ | $\frac{1}{2} \text{ M}$ |
| | | $\frac{1}{2} \text{ M}$ |
| | | |
| | | $\frac{1}{2} \text{ M}$ |

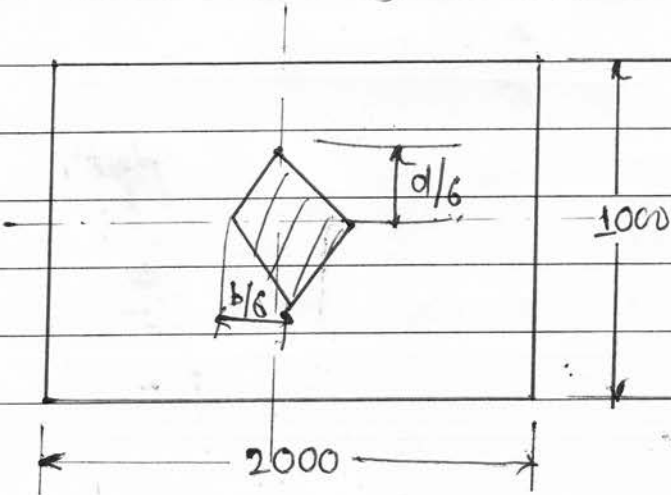
| Q.NO | SOLUTION | MARKS |
|------|---|-------|
| | $M = M_{yp} = Pe$ $= 150 \times 10^3 \times 50$ $M = 7.5 \times 10^6 \text{ N-mm}$ | 1/2 M |
| | $Z = I_{yp}$ $= \frac{db^2}{6}$ $= \frac{100 \times (150)^2}{6}$ $Z = 375 \times 10^3 \text{ mm}^3$ | 1/2 M |
| | $\sigma_b = \frac{M}{Z} = \frac{M_{yp}}{I_{yp}}$ $= \frac{7.5 \times 10^6}{375 \times 10^3}$ $\sigma_b = 20 \text{ N/mm}^2$ | 1/2 M |
| | $\therefore \sigma_{max} = \sigma_0 + \sigma_b$ $= 10 + 20$ $\sigma_{max} = 30 \text{ N/mm}^2 \rightarrow \text{Compressive Nature.}$ | 1/2 M |
| | $\sigma_{min} = \sigma_0 - \sigma_b$ $= 10 - 20$ $\sigma_{min} = -10 \text{ N/mm}^2 \text{ i.e. tensile nature}$ | 1/2 M |

| Q.NO | SOLUTION | MARKS |
|------|---|-----------------|
| 5-d) | Given:- for circular column | |
| | $D = \text{External diameter} = 40 \text{ cm} = 400 \text{ mm}$ | |
| | $d = \text{Internal diameter} = 30 \text{ cm} = 300 \text{ mm}$ | |
| | $P = 150 \times 10^3 \text{ N} = \text{load}$ | |
| | To find:- (i) σ_{max} | |
| | (ii) σ_{min} | |
| | sol ⁿ :- We know that | |
| | | |
| | | |
| | | |
| | | |
| | | |
| | | |
| | | |
| | $\sigma_o = \frac{P}{A} = \frac{150 \times 10^3}{\frac{\pi}{4} \times (400^2 - 300^2)}$ | $\frac{1}{2} M$ |
| | $\sigma_o = 2.7283 \text{ N/mm}^2$ | $\frac{1}{2} M$ |
| | | |
| | | |
| | $\sigma_b = \frac{M}{Z} = \frac{M}{I/Y}$ | $\frac{1}{2} M$ |
| | $= \frac{M Y}{I}$ | |
| | $= \frac{P \times e \times D/2}{\frac{\pi}{64} \times (D^4 - d^4)}$ | $\frac{1}{2} M$ |
| | $= \frac{150 \times 10^3 \times 200 \times (400/2)}{\frac{\pi}{64} \times (40^4 - 30^4)}$ | $\frac{1}{2} M$ |
| | | |
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| Q.NO | SOLUTION | MARKS |
|------|---|----------------------------------|
| | $\sigma_b = \frac{150 \times 10^3 \times 200 \times 200}{859.029 \times 10^6}$ | |
| | $\sigma_b = 6.9846 \text{ N/mm}^2$ | 1/2 M |
| | $\begin{aligned} \sigma_{\max} &= \sigma_0 + \sigma_b \\ &= 2.728 + 6.984 \\ \sigma_{\max} &= 9.7126 \text{ N/mm}^2 \end{aligned}$ | 1/2 M → compressive in Nature |
| | $\begin{aligned} \sigma_{\min} &= \sigma_0 - \sigma_b \\ &= 2.728 - 6.984 \\ \sigma_{\min} &= -4.256 \text{ N/mm}^2 \end{aligned}$ | 1/2 M → Tensile in Nature. |
| 5-e) | Given:- for a rectangular rod bent into | |
| | <p>y 'C' shape.</p>  <p>The diagram shows a C-shaped rod with a total width of 300 mm and a height of 50 mm. A vertical load of 40 kN is applied to the top flange. A coordinate system (x, y) is shown with the x-axis horizontal and the y-axis vertical.</p> | |

| Q.NO | SOLUTION | MARKS |
|-----------|--|-------------------------|
| 5-e | $b = 100 \text{ mm}$ | |
| Continues | $d = 50 \text{ mm}$ | |
| | $P = 40 \text{ kD} = 40 \times 10^3 \text{ N}$ | |
| | $e = 300 \text{ mm}$ | |
| | To find:- Resultant along X-X. | |
| | solution:- $\sigma_{\text{max}} = \sigma_0 + \sigma_b$ | |
| | $\sigma_{\text{min}} = \sigma_0 - \sigma_b$ | |
| | $\sigma_0 = \frac{P}{A} = \frac{40 \times 10^3}{(100 \times 50)}$ | |
| | $\sigma_0 = 8 \text{ N/mm}^2$ | $\frac{1}{2} \text{ M}$ |
| | $M = Pe$ | |
| | $= 40 \times 10^3 \times 300$ | |
| | $M = 12 \times 10^6 \text{ mm}^3$ | $\frac{1}{2} \text{ M}$ |
| | $y = \frac{100}{2} = 50 \text{ mm}$ | |
| | $I = \frac{db^3}{12} = \frac{50 \times (100)^3}{12}$ | $\frac{1}{2} \text{ M}$ |
| | $I = 4.166 \times 10^6 \text{ mm}^4$ | $\frac{1}{2} \text{ M}$ |
| | $\sigma_b = \frac{My}{I} = \frac{12 \times 10^6 \times 50}{4.166 \times 10^6}$ | $\frac{1}{2} \text{ M}$ |
| | $\sigma_b = 144 \text{ N/mm}^2$ | $\frac{1}{2} \text{ M}$ |

| Q.NO | SOLUTION | MARKS |
|------|--|-------------------------|
| | $\sigma_{\max} = \sigma_a + \sigma_b$ $= 8 + 144$ $= 152 \text{ MPa}$ | |
| | <div style="border: 1px solid black; display: inline-block; padding: 2px;">$\sigma_{\max} = 152 \text{ MPa}$</div> \longrightarrow 'Tension' | $\frac{1}{2} \text{ M}$ |
| | $\sigma_{\min} = \sigma_a - \sigma_b$ $= 8 - 144$ $\sigma_{\min} = -136$ | |
| | <div style="border: 1px solid black; display: inline-block; padding: 2px;">$\sigma_{\min} = -136 \text{ MPa}$</div> \longrightarrow compression | $\frac{1}{2} \text{ M}$ |
| | <p>As rod is bent into 'c' shape, it is tension member so tension is considered as +ve and compression is taken as -ve.</p> | |
| | <p>Resultant stresses developed at section x-x.</p> $\sigma_{\max} = 152 \text{ MPa}$ $\sigma_{\min} = 136 \text{ MPa}$ | |
| 5-f) | <p>Given:- for rectangular cross section</p>  | |
| | | 01M |

| Q.NO | SOLUTION | MARKS |
|------|---|-------|
| | To find:- Limit of eccentricity | |
| | <u>solution:- Case-I</u> :- load along x-axis | |
| | $\sigma_o = \sigma_b$ | |
| | $\frac{P}{bd} = \frac{P \times e}{db^2/6}$ | 1/2 M |
| | $\frac{P}{bd} = \frac{6Pe}{db^2}$ | |
| | $\frac{P}{bd}$ = $\frac{6Pe}{db^2}$ | |
| | $1 = \frac{6e}{b}$ | |
| | $e = b/6$ | 1/2 M |
| | $= 1000/6$ | |
| | $e = 166.66 \text{ mm}$ → along x-axis | 1/2 M |
| | Case-II :- load along y-axis | |
| | $\sigma_o = \sigma_b$ | |
| | $\frac{P}{bd} = \frac{Pe}{bd^2/6}$ | 1/2 M |
| | $\frac{P}{bd} = \frac{6Pe}{bd^2}$ | |
| | $\frac{P}{bd}$ = $\frac{6Pe}{bd^2}$ | |
| | $1 = \frac{6e}{d} \quad \therefore e = d/6$ | 1/2 M |
| | $e = 2000/6 = 333.33 \text{ mm}$ → Along y-axis | 1/2 M |

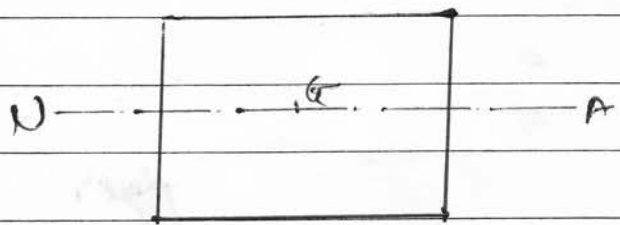
| Q.NO | SOLUTION | MARKS |
|------|---|-------------------------|
| 6-a) | Assumptions of Pure torsion are, | |
| | 1) The shaft material is homogeneous & isotropic | (1 M for each any four) |
| | 2) The shaft is straight having uniform cross-section | |
| | 3) The shaft is acted upon by pure torsion | |
| | 4) shear stress is directly proportional to shear strain. | |
| | 5) section plane before twisting moment will remain plane even after application of twisting moment. | |
| 6-b) | <u>Given</u> :- for a shaft, used to transmit power $P = 20 \text{ kW} = 20 \times 10^3 \text{ W}$ $N = 150 \text{ rpm}$ $T_{\text{max}} = \text{Maximum torque} = 40\% \text{ of } T_{\text{ave}}$ $= 0.4 T_{\text{ave}}$ $T_{\text{max}} = 1.4 T_{\text{ave}}$ | $\frac{1}{2} \text{ M}$ |
| | $\tau = 50 \text{ MPa} = 50 \text{ N/mm}^2$ | |
| | <u>To find</u> :- diameter of shaft | |
| | <u>solution</u> :- $P = \frac{2\pi P T_{\text{ave}}}{60}$ | $\frac{1}{2} \text{ M}$ |
| | $20 \times 10^3 = \frac{2\pi \times 150 \times T_{\text{ave}}}{60}$ | |
| | $T_{\text{ave}} = 1.273 \times 10^3 \text{ N}\cdot\text{m}$ | |
| | $T_{\text{ave}} = 1.273 \times 10^6 \text{ N}\cdot\text{mm}$ | $\frac{1}{2} \text{ M}$ |

| Q.NO | SOLUTION | MARKS |
|-------|--|----------------|
| | $T_{max} = 1.4 T_{ave}$ | |
| | $T_{max} = 1.4 \times 1.273 \times 10^6$ | |
| | $T_{max} = 1.782 \times 10^6 \text{ Nm}$ | 01M |
| | $1.782 \times 10^3 = \frac{\pi}{16} \tau \times D^3$ | $\frac{1}{2}M$ |
| | $= \frac{\pi}{16} \times 50 \times D^3$ | |
| | $D = 56.63 \text{ mm} \rightarrow \text{Ans}$ | 01M |
| Q-6c) | Given:- for a shaft | |
| | $d = 40 \text{ mm} = \text{diameter of shaft}$ | |
| | $N = 2000 \text{ rpm}$ | |
| | $\tau = \text{shear stress} = 85 \text{ MPa}$ | |
| | $= 85 \text{ N/mm}^2$ | |
| | To find :- Power transmitted by shaft | |
| | <u>Solution:-</u> $P = \frac{2\pi NT}{60}$ | $\frac{1}{2}M$ |
| | $T = \frac{\pi}{16} \tau \times d^3$ | $\frac{1}{2}M$ |
| | $= \frac{\pi}{16} \times 85 \times (40)^3$ | |
| | $= 1.068 \times 10^6 \text{ Nmm}$ | |
| | $T = 1.068 \times 10^3 \text{ Nm}$ | 01M |
| | $P = \frac{2\pi \times 2000 \times 1.068 \times 10^3}{60}$ | 01M |
| | $P = 22.37 \times 10^3 \text{ W}$ | |
| | $P = 22.37 \text{ kW} \rightarrow \text{Ans}$ | 01M |

| Q.NO | SOLUTION | MARKS |
|------|---|-------------------------|
| 6-d) | <u>Given</u> :- for a shaft | |
| | $P = \text{Power transmitted} = 150 \text{ kW}$ | |
| | $P = 150 \times 10^3 \text{ W}$ | |
| | $N = 2000 \text{ rpm}$ | |
| | $\tau = \text{shear stress} = 80 \text{ N/mm}^2$ | |
| | $\alpha = \text{Angle of twist} = 1.5^\circ$ | |
| | $= \frac{1.5 \times \pi}{180} = 0.02167 \text{ radians.}$ | |
| | $L = \text{Length of shaft} = 4 \text{ m}$ | |
| | $= 4000 \text{ mm}$ | |
| | $G = \text{Modulus of rigidity}$ | |
| | $= 0.8 \times 10^5 \text{ N/mm}^2$ | |
| | <u>To find</u> :- Diameter of shaft | |
| | <u>solution</u> :- (1) on the basis of strength | |
| | $T = \frac{\pi}{16} \tau D^3$ | $\frac{1}{2} \text{ M}$ |
| | $P = \frac{2\pi NT}{60}$ | $\frac{1}{2} \text{ M}$ |
| | $150 \times 10^3 = \frac{2\pi \times 2000 \times T}{60}$ | |
| | $T = 7.16 \times 10^3 \text{ Nm}$ | $\frac{1}{2} \text{ M}$ |
| | $T = 7.16 \times 10^6 \text{ Nmm}$ | |
| | $7.16 \times 10^6 = \frac{\pi}{16} \times 80 \times D^3$ | $\frac{1}{2} \text{ M}$ |
| | $D = \frac{(7.16 \times 10^6)}{(\frac{\pi}{16} \times 80)} = 76.9669$ | $\frac{1}{2} \text{ M}$ |
| | $D = 76.97 \text{ mm}$ | $\frac{1}{2} \text{ M}$ |

| Q.NO | SOLUTION | MARKS |
|------|---|-------|
| | <p>② On the basis of angle of twist i.e. stiffness</p> | |
| | $\frac{T}{J} = \frac{G\theta}{L}$ | 1/2 M |
| | $\frac{T}{\pi/32 \times D^4} = \frac{G\theta}{L}$ | |
| | $\frac{7.16 \times 10^6}{\frac{\pi}{32} \times D^4} = \frac{0.8 \times 10^5 \times 0.0261}{4000}$ | |
| | $D^4 = \frac{7.16 \times 10^6 \times 32 \times 4000}{0.8 \times 10^3 \times 0.0261 \times \pi}$ | 1/2 M |
| | $= \frac{9.1648 \times 10^{11}}{6.5596 \times 10^3}$ | |
| | $D^4 = 139.71 \times 10^6$ | |
| | $D = 108.72 \text{ mm}$ | 1/2 M |
| | <p>larger of above two values, for diameter is taken as suitable diameter</p> | |
| | $\therefore D = 108.72 \text{ mm} \quad \longrightarrow \text{Ans.}$ | |

| Q.NO | SOLUTION | MARKS |
|------|--|-------|
| 6c) | <p><u>Given:-</u> $D = \text{External diameter} = 400 \text{ mm}$ $d = \text{Internal diameter} = 200 \text{ mm}$ $T = \text{Twisting moment} = 4650 \text{ Nm}$ $= 4650 \times 10^3 \text{ N-mm}$ $G = \text{Modulus of rigidity} = 82 \times 10^3 \text{ N/mm}^2$ $L = 20D$ $= 20 \times 400 = 8000 \text{ mm}$</p> | |
| | <p><u>To find</u> (i) Maximum intensity of stress (ii) Angle of twist</p> | |
| | <p><u>Solution:-</u> (i) Maximum intensity of stress</p> | |
| | $\frac{T}{J} = \frac{T_{\text{max}}}{R}$ | 1/2 M |
| | $J = \frac{\pi}{32} \times (D^4 - d^4)$ | |
| | $= \frac{\pi}{32} \times (400^4 - 200^4)$ | |
| | $J = 2.3561 \times 10^9 \text{ mm}^4 \rightarrow \text{Polar M.I.}$ | 1/2 M |
| | $R = D/2 = \frac{400}{2} = 200 \text{ mm}$ | 1/2 M |
| | $\frac{4650 \times 10^3}{2.3561 \times 10^9} = \frac{T_{\text{max}}}{200}$ | |
| | $T_{\text{max}} = \frac{4650 \times 10^3 \times 200}{2.3561 \times 10^9}$ | |
| | $T_{\text{max}} = 0.3947 \text{ N/mm}^2$ | 1/2 M |

| Q.NO | SOLUTION | MARKS |
|------|---|----------------|
| | ① Angle of twist | |
| | $\frac{T}{J} = \frac{G\theta}{L}$ | $\frac{1}{2}M$ |
| | $\frac{4650 \times 10^3}{2.3561 \times 10^9} = \frac{82 \times 10^3 \times \theta}{8000}$ | |
| | $\theta = \frac{4650 \times 10^3 \times 8000}{2.3561 \times 10^9}$ | $\frac{1}{2}M$ |
| | $= 1.9254 \times 10^6$ | |
| | $\theta = 1.9254 \times 10^6 \text{ radians}$ | |
| | $= 1.9254 \times \frac{180}{\pi}$ | |
| | $\theta = 0.011^\circ$ | 01M |
| | 6-f-i) Neutral Axis | |
| | <p>- When beam is subjected to pure bending there is a layer which will not ^{neither} be subjected to tension nor compression, such a layer is called neutral layer or neutral axis (N-A)</p> | |
| |  | |
| | <p>- Neutral Axis always passes through centroid or C.G.</p> | 02M |

